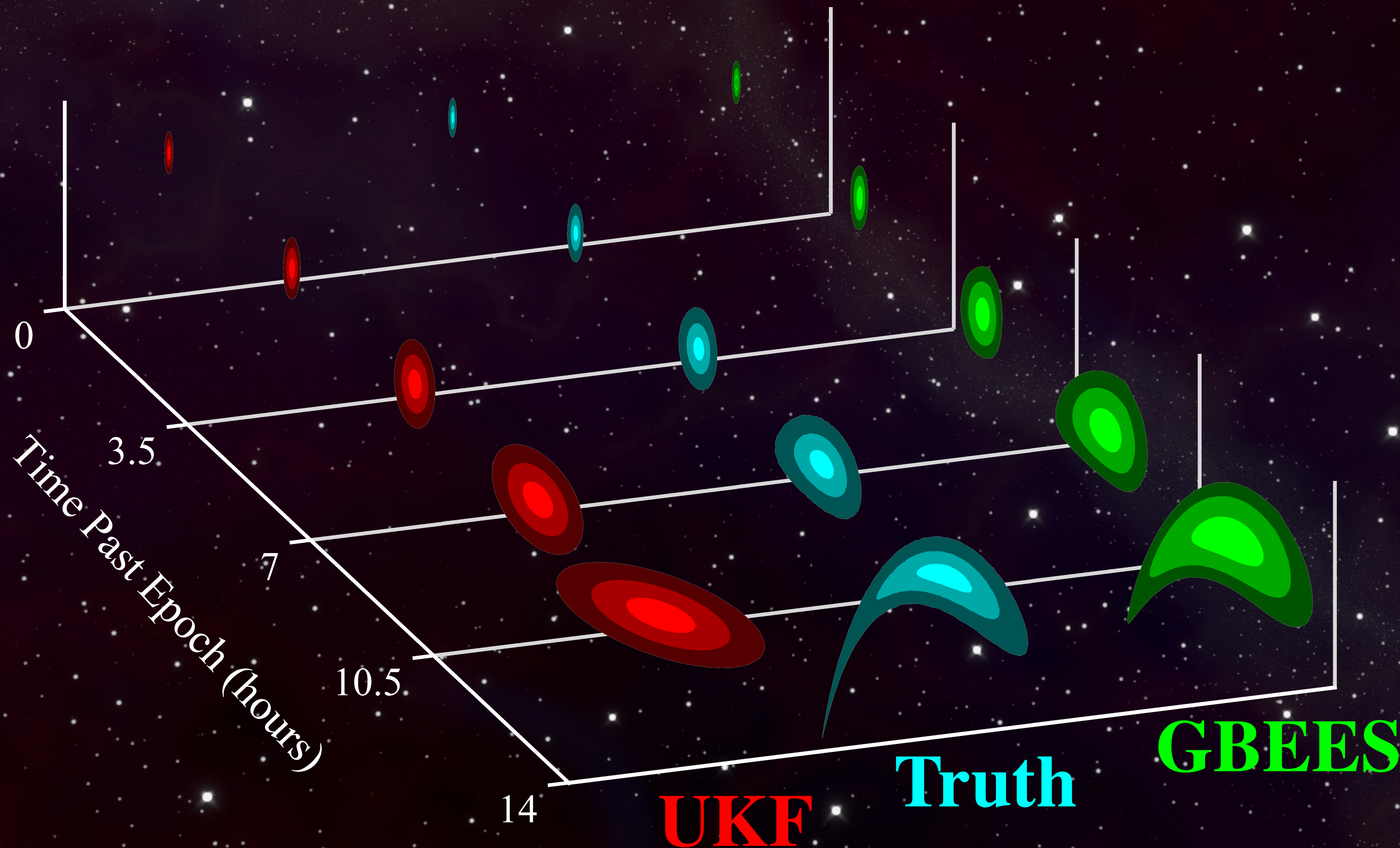




# ON THE VALIDITY OF THE GAUSSIAN ASSUMPTION IN THE JOVIAN SYSTEM: EVALUATING LINEAR AND NONLINEAR FILTERS FOR MEASUREMENT-SPARSE ESTIMATION

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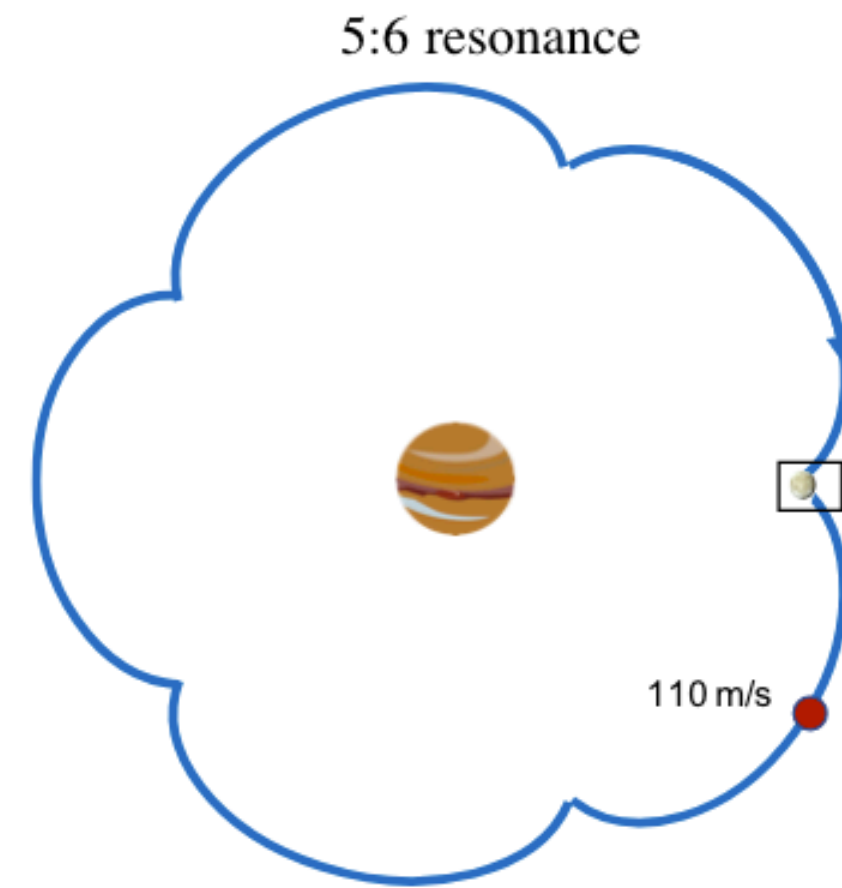
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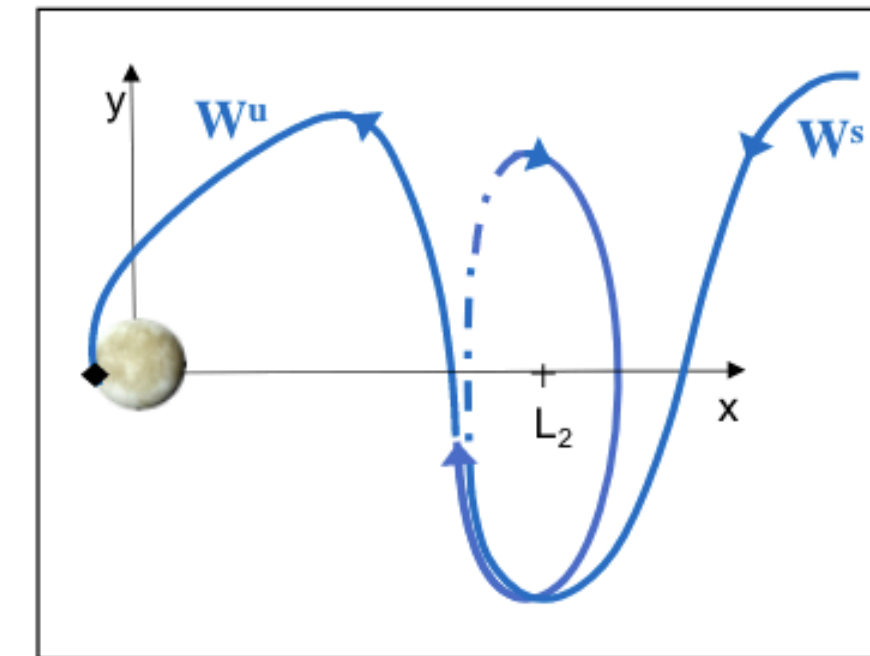
## Time validity of the Gaussian assumption of uncertainty

- A theoretically  $\Delta V$ -free, ballistic capture of a Europa lander realistically requires **statistical maneuvers** to maintain Gaussian error in position and velocity for linearized navigation techniques

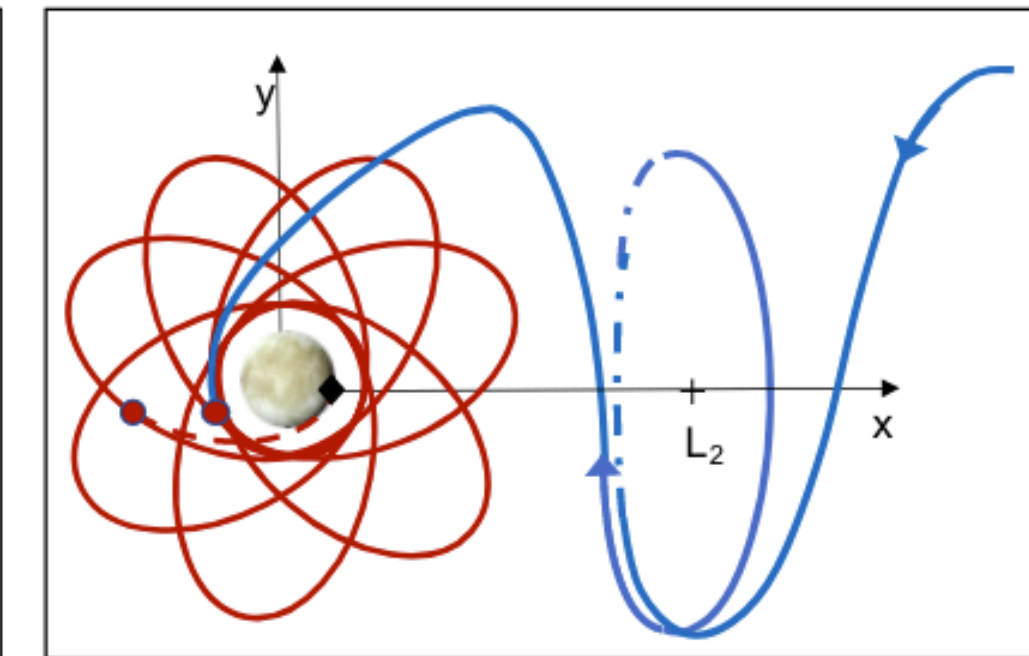
**Proposed  $\Delta V$ -free ballistic capture**



a) Direct landing

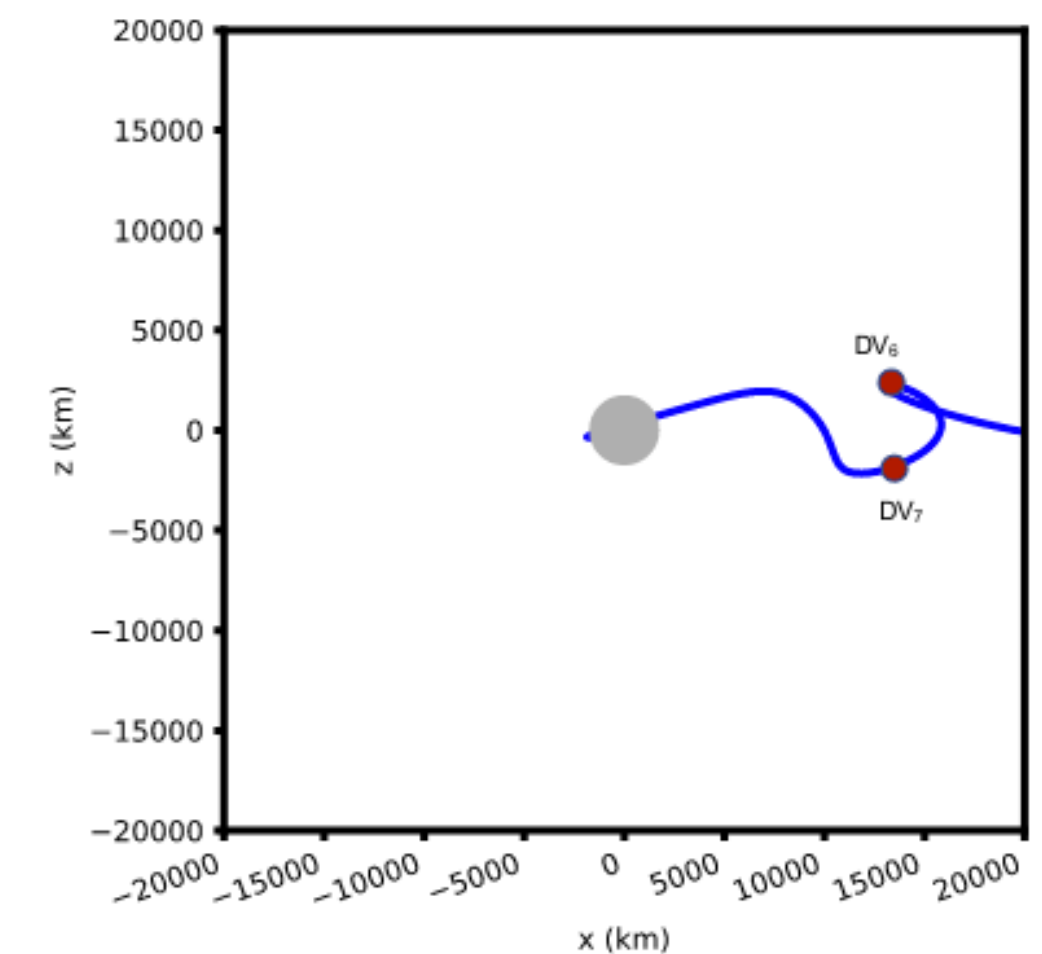
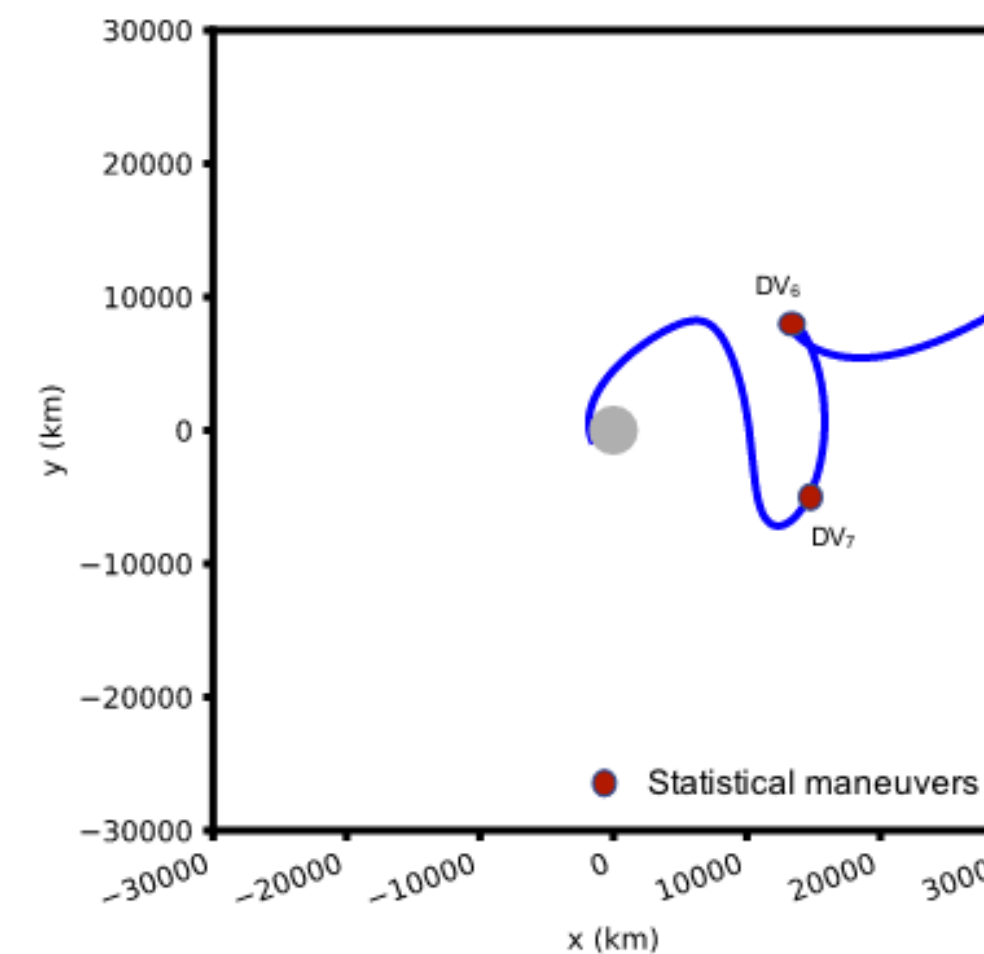
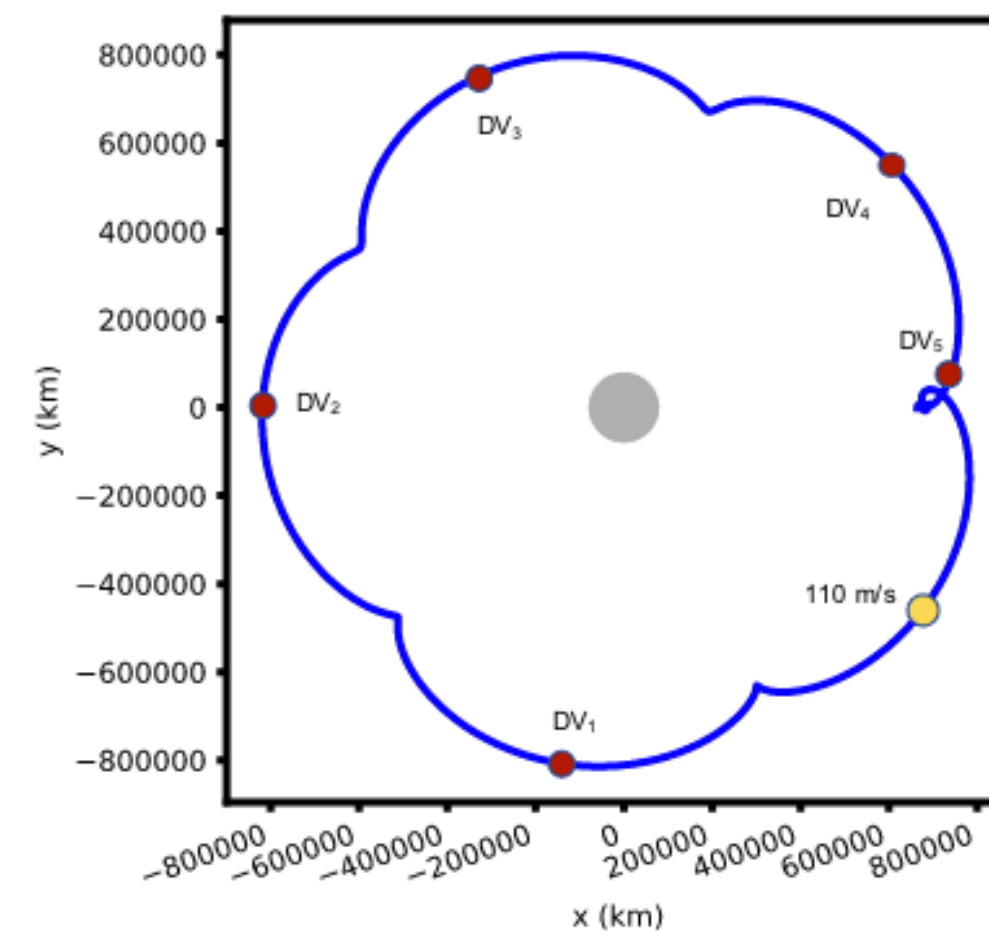


b) Capture, with optional landing



— Manifold    - - - Optional staging orbit    — Capture orbit  
 ● Maneuver    ◆ Landing site

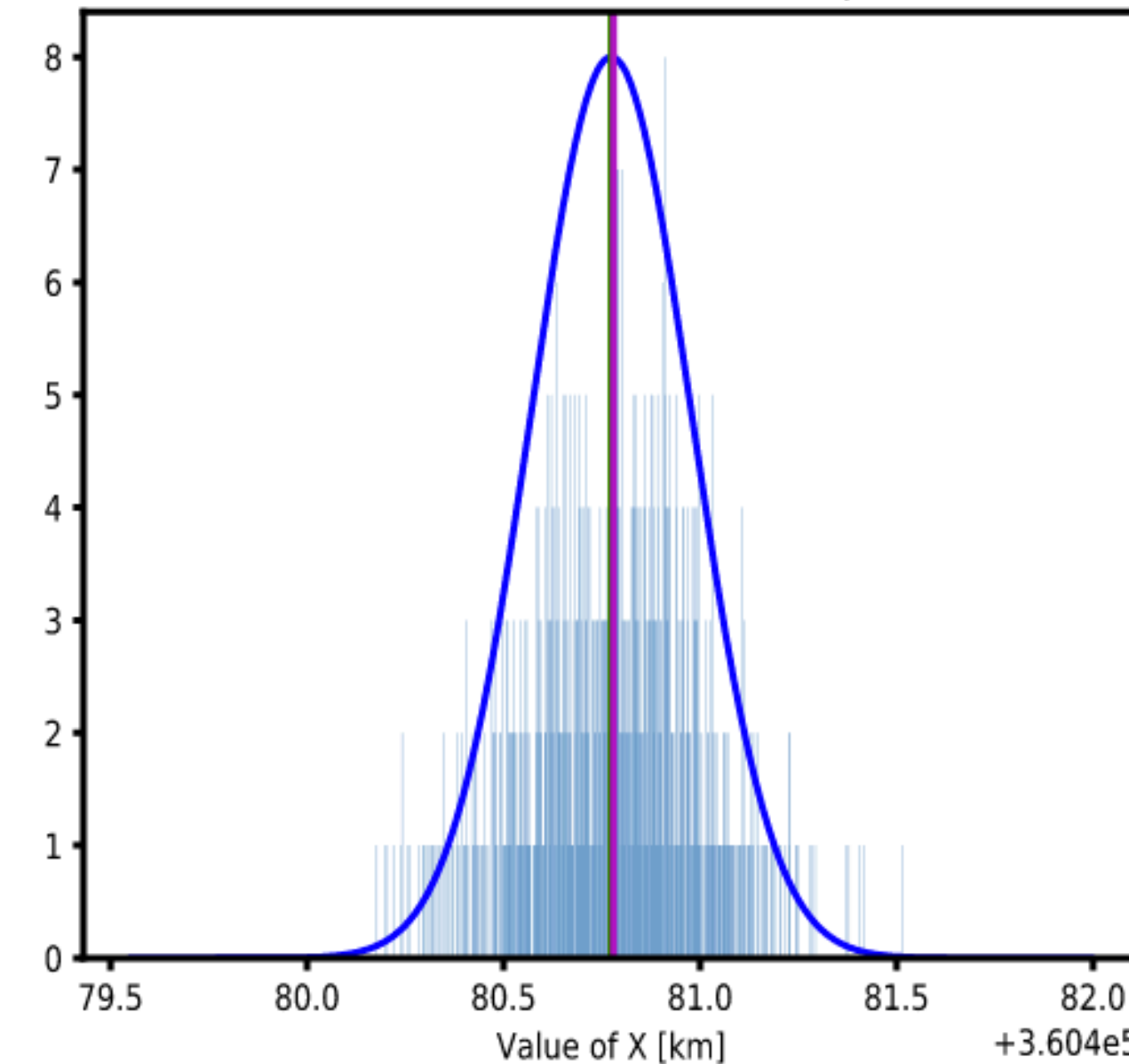
**Actual trajectory with statistical maneuvers  $\Delta V_i$**



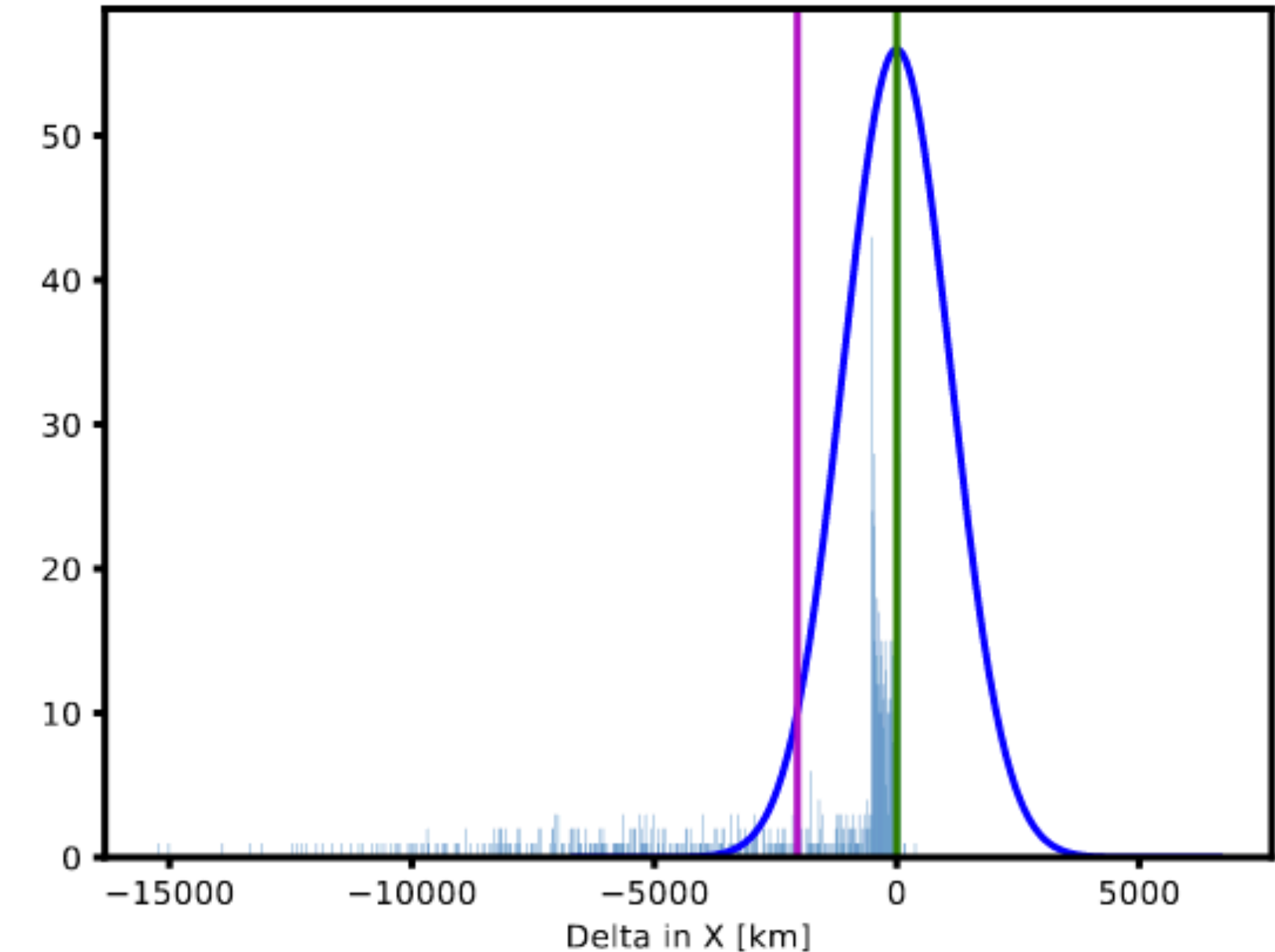
## Time validity of the Gaussian assumption of uncertainty

- A theoretically  $\Delta V$ -free, ballistic capture of a Europa lander realistically requires **statistical maneuvers** to maintain Gaussian error in position and velocity for linearized navigation techniques

### Proposed $\Delta V$ -free ballistic capture



Initial Gaussian uncertainty at leveraging maneuver

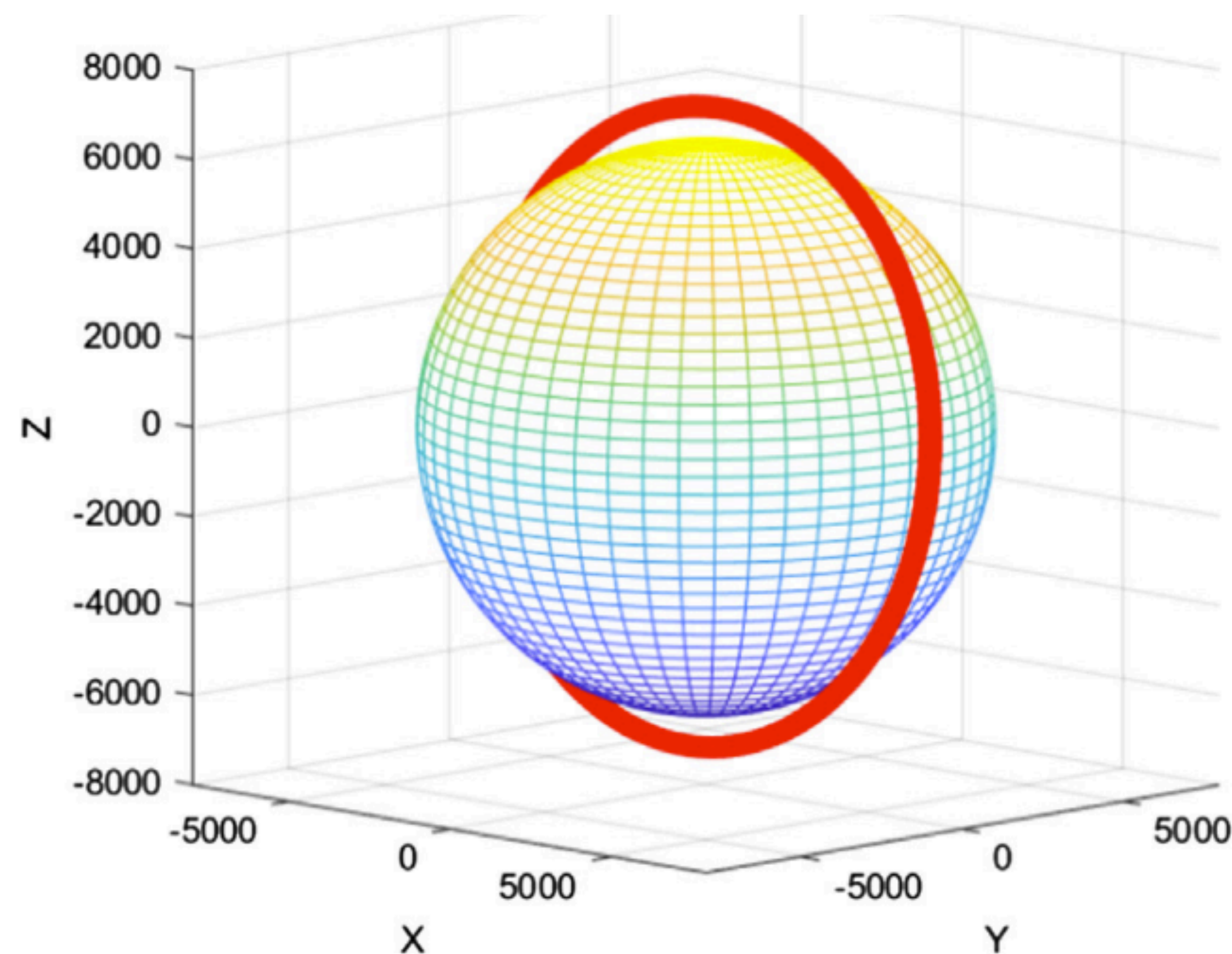


Final non-Gaussian uncertainty at Europa arrival

**KEY QUESTION:** What are the temporal limits of linear filters in the Jovian regime, and when might it be necessary to implement nonlinear filters?



- Previous work has focused on the efficacy of linear/nonlinear filters applied to LEO trajectories in measurement-sparse conditions
  - \* Initial condition resulting in highly-inclined, nearly-circular LEO
  - \* Propagated for 6 revolutions (4.94 hours) w/ RK8(7)
  - \* Negligible process noise ( $Q = 0$ )



$$\mathbf{x}_0 = \begin{bmatrix} a \text{ (km)} \\ e \text{ ( )} \\ i \text{ (}^\circ\text{)} \\ \Omega \text{ (}^\circ\text{)} \\ \omega \text{ (}^\circ\text{)} \\ M \text{ (}^\circ\text{)} \end{bmatrix} = \begin{bmatrix} 7,078.0068 \\ 0.01 \\ 85^\circ \\ 0^\circ \\ 0^\circ \\ 0^\circ \end{bmatrix}$$

$$\sigma_r = 30 \text{ m}, \quad \sigma_v = 0.3 \text{ m/s}$$

Dynamic Model	Description
Primary Body Gravity Third-Body Perturbations Atmospheric Drag Solar Radiation Pressure	70 x 70 Sun and Moon Cannonball Cannonball



## Truth Model

- We plan to implement a similar framework as previously shown, applied to trajectories in the Jovian system, with the following important changes implemented:

### I. Truth Model

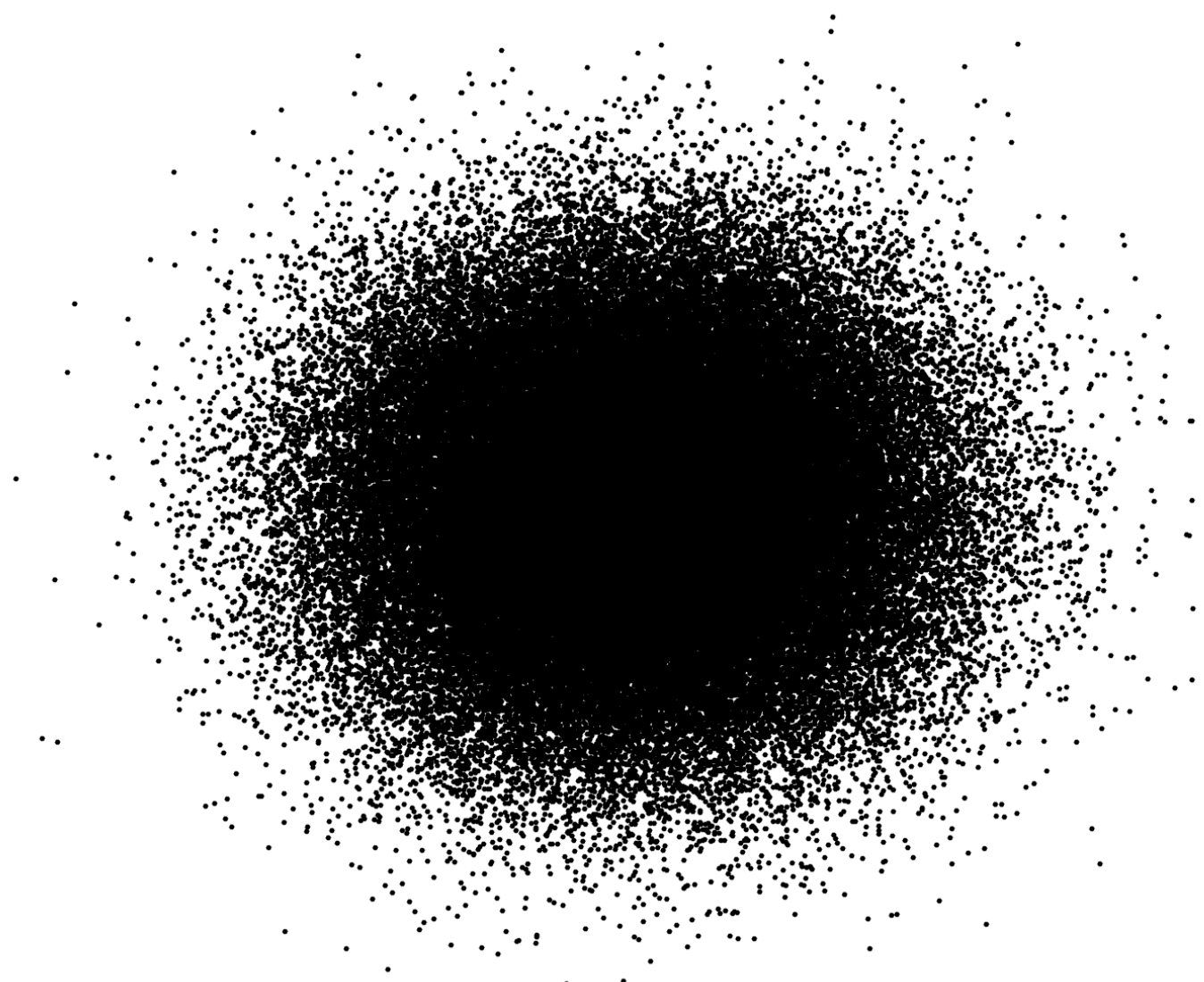
- \* A **high-resolution particle filter** will allow for **confidence interval** comparison with linear filters, providing more information than a high-resolution Monte Carlo distribution

#### For Monte Carlo/Particle Filter:

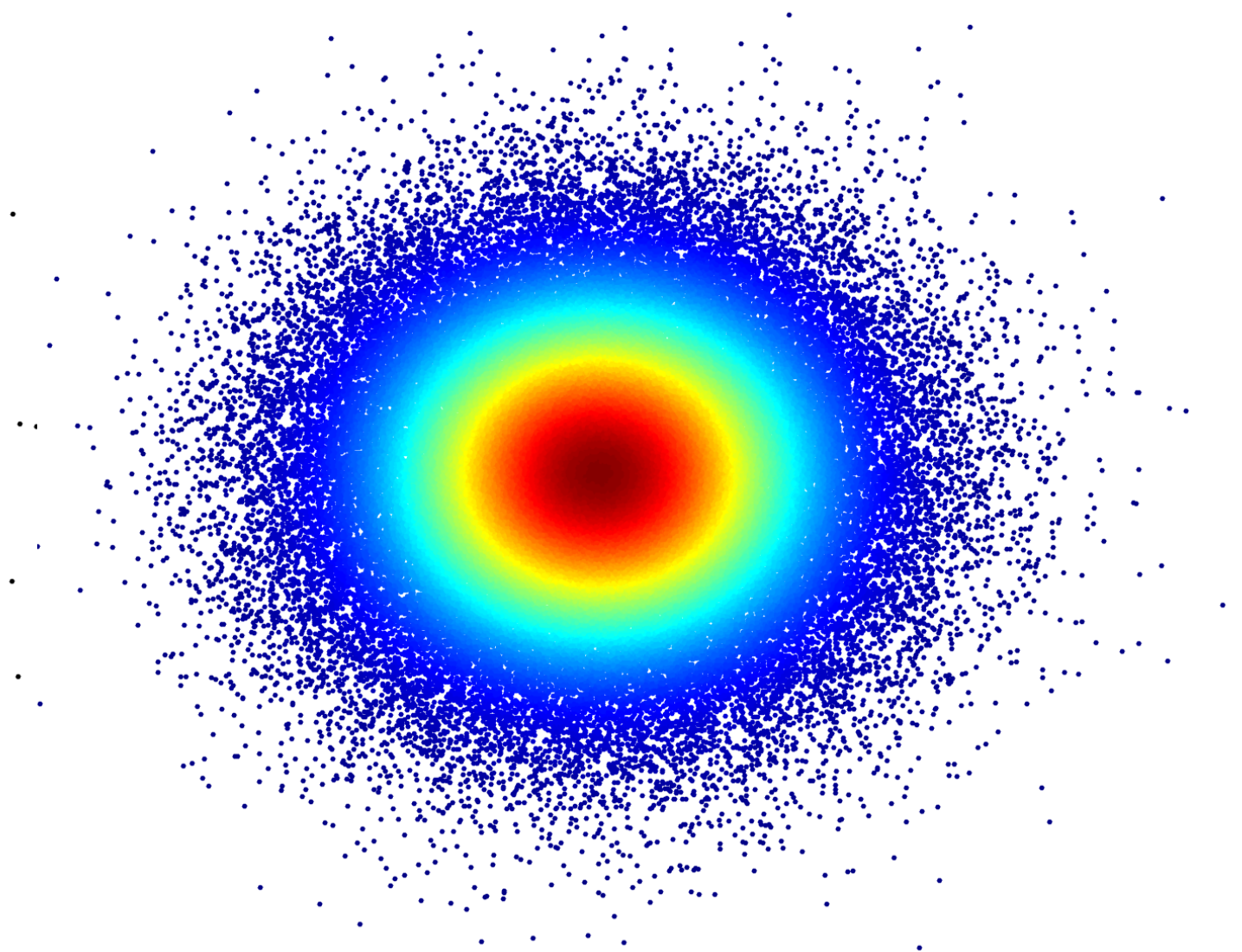
$$\{x\} \sim \mathcal{N}(\mu, \Sigma)$$

#### For Particle Filter only:

$$\{p(x)\} \sim \exp \left\{ -\frac{1}{2}(x - \mu)^T \Sigma^{-1}(x - \mu) \right\}$$



Monte Carlo Interpretation



Particle Filter Interpretation



- We plan to implement a similar framework as previously shown, applied to trajectories in the Jovian system, with the following important changes implemented:

## 2. Distribution Comparison Metric

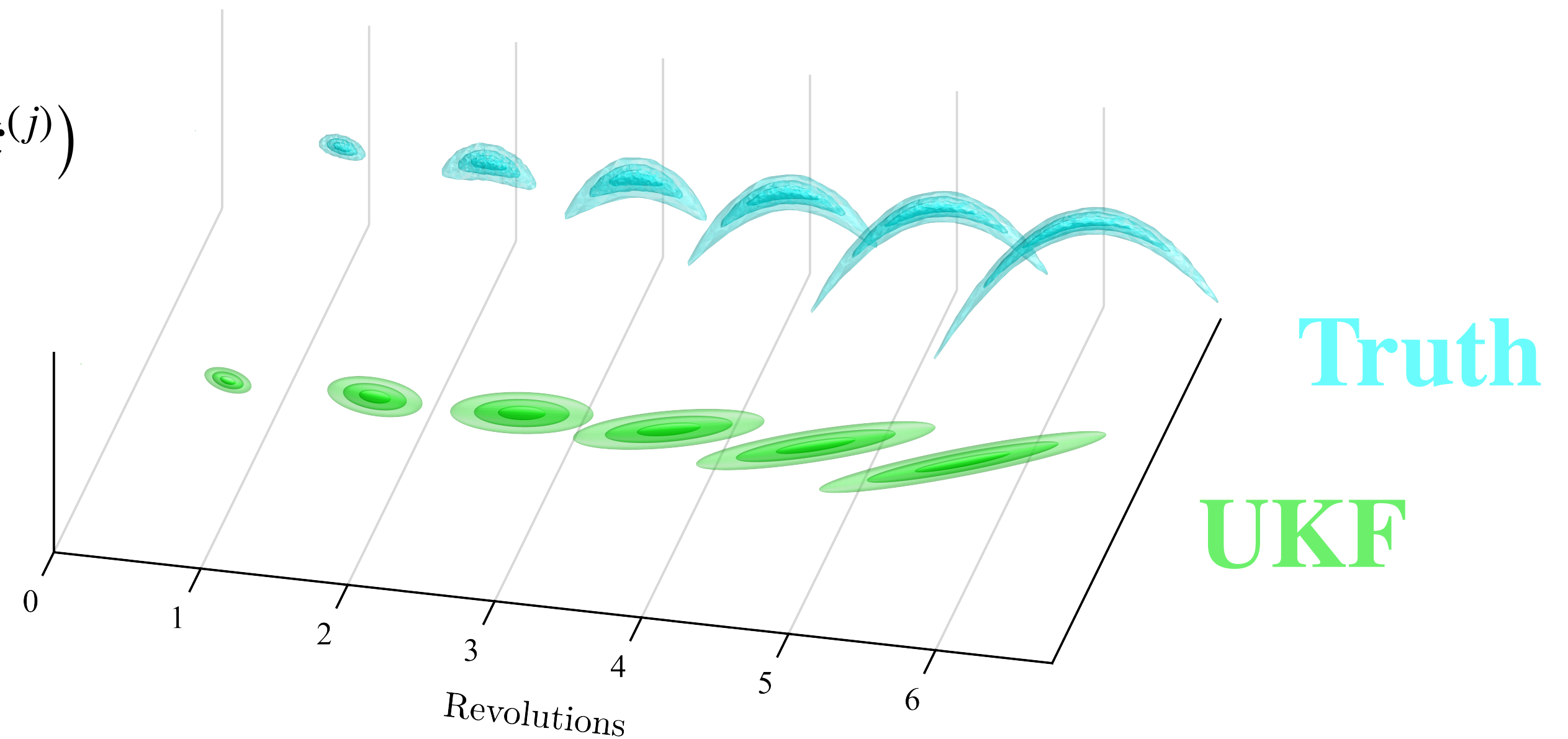
- \* The metric used to indicate diverge (previously SNEES) should consider **the true probability distribution is non-Gaussian** after enough propagation time without measurements

$$SNEES = \frac{1}{Md} \sum_{j=1}^M (\mathbf{x}^{(j)} - \hat{\mathbf{x}}^{(j)})^T (\hat{\Sigma}^{(j)})^{-1} (\mathbf{x}^{(j)} - \hat{\mathbf{x}}^{(j)})$$

- ◆ **Problem:** Assumes Gaussian errors

$$D_{KL}(P || Q) = \sum_{x \in \mathcal{X}} P(x) \log \left( \frac{P(x)}{Q(x)} \right)$$

- ◆ **Problem:** Diverges for extremely low probability events when distributions differ



Gaussian uncertainty propagated with Two-Body Dynamics becoming highly non-Gaussian

\*  $SNEES$  : Scaled Normalized Estimation Error Squared

\*  $D_{KL}$  : Kullback-Leibler Divergence





# Jovian Application: Framework Changes

## Propagation Conditions

- We plan to implement a similar framework as previously shown, applied to trajectories in the Jovian system, with the following important changes implemented:

### 3. Propagation Conditions

- \* To test the limits of the linear filters, we plan on performing **“measurementless”** propagation
  - ◆ We consider negligible process noise ( $Q = 0$ ) and correct initial measurements ( $\delta\mathbf{x}_0 = \mathbf{x}_0 - \hat{\mathbf{x}}_0 = 0$ )
- \* Purely two-body dynamics will be propagated, so the following results are likely a **best-case scenario**

$$\mathbf{x} = \left[ a, e, i, \Omega, \omega, M \right]^T, \quad \dot{\mathbf{x}} = \left[ 0, 0, 0, 0, 0, \sqrt{\frac{\mu}{a^3}} \right]^T$$

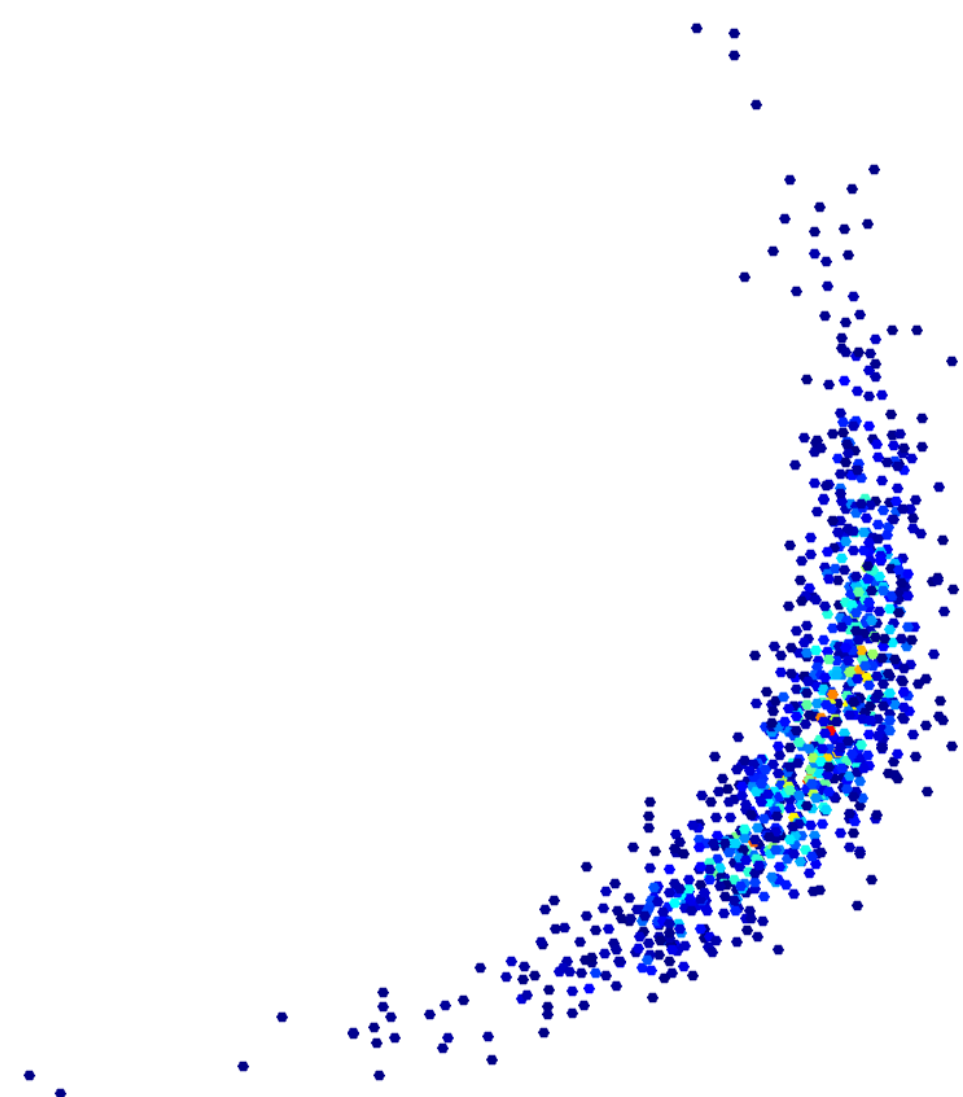
- ◆ Future work will aim to feed the dynamics from an **ephemeris-level numerical propagator**
- \* Filter parameters:

Filter	Parameters
Particle Filter (truth)	Particles: $10^5$
UKF	$\alpha = 10^{-3}, \beta = 2, \kappa = 0$
EnKF	Members: $10^4$

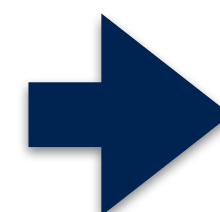


## Choosing a metric for Gaussian/non-Gaussian distribution comparison

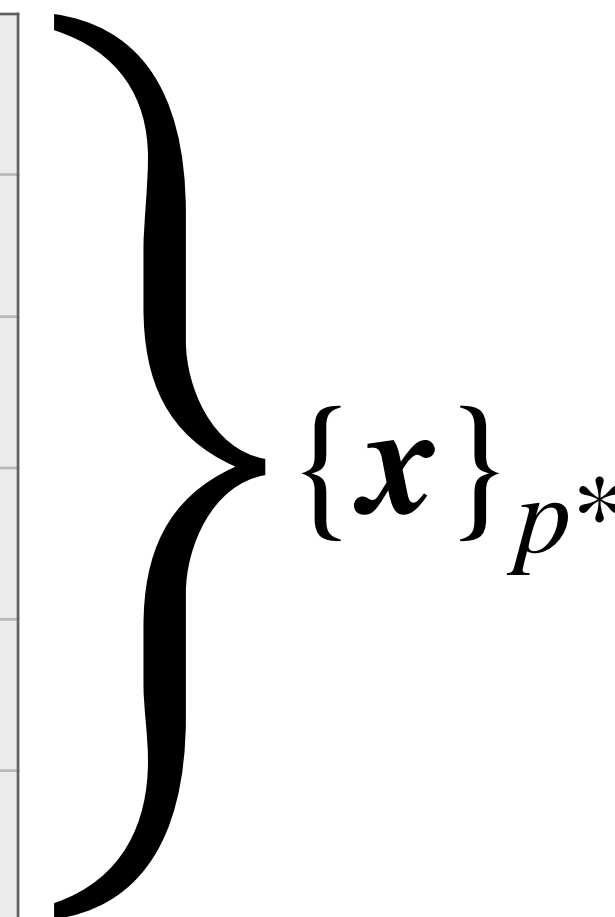
Point Mass  
Representation



●	$p_1$	$[x_1, x_2, \dots, x_d]_1$
●	$p_2$	$[x_1, x_2, \dots, x_d]_2$
●	$p_3$	$[x_1, x_2, \dots, x_d]_3$
●	$p_4$	$[x_1, x_2, \dots, x_d]_4$
●	$p_5$	$[x_1, x_2, \dots, x_d]_5$
●	$p_n$	$[x_1, x_2, \dots, x_d]_n$



●	$p_1$	$[x_1, x_2, \dots, x_d]_1$
●	$p_2$	$[x_1, x_2, \dots, x_d]_2$
●	$p_3$	$[x_1, x_2, \dots, x_d]_3$
●	$p_4$	$[x_1, x_2, \dots, x_d]_4$
●	$p_5$	$[x_1, x_2, \dots, x_d]_5$
⋮		⋮
●	$p_n$	$[x_1, x_2, \dots, x_d]_n$

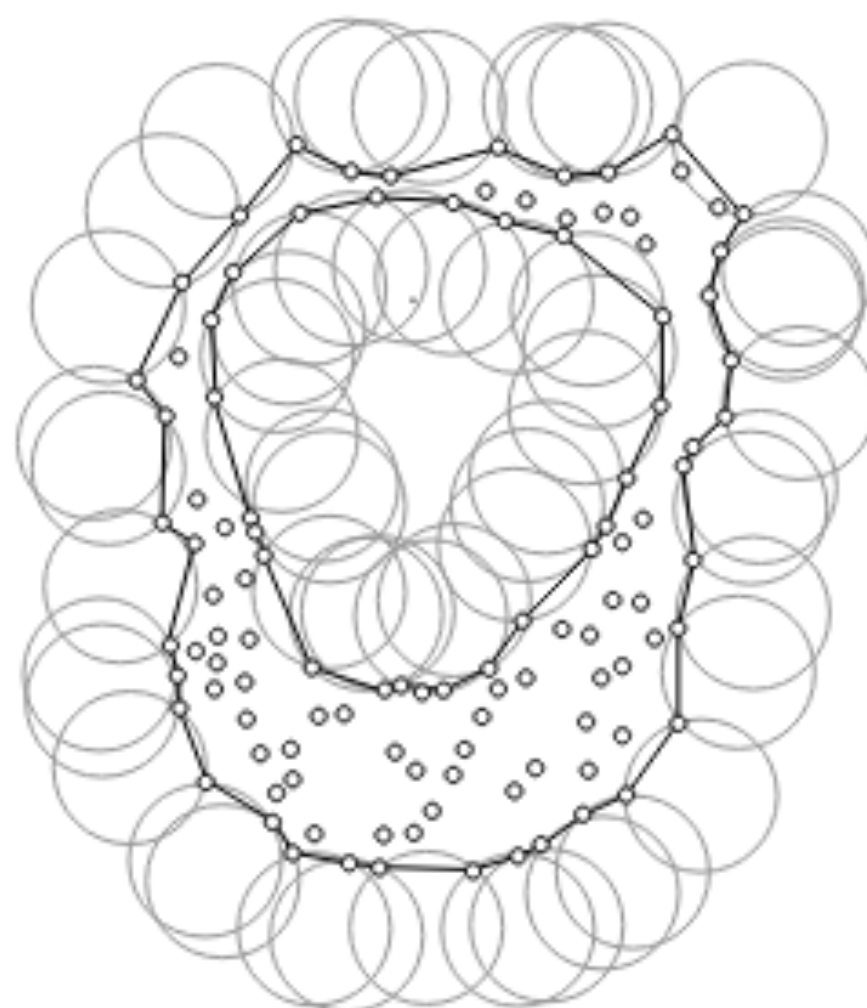
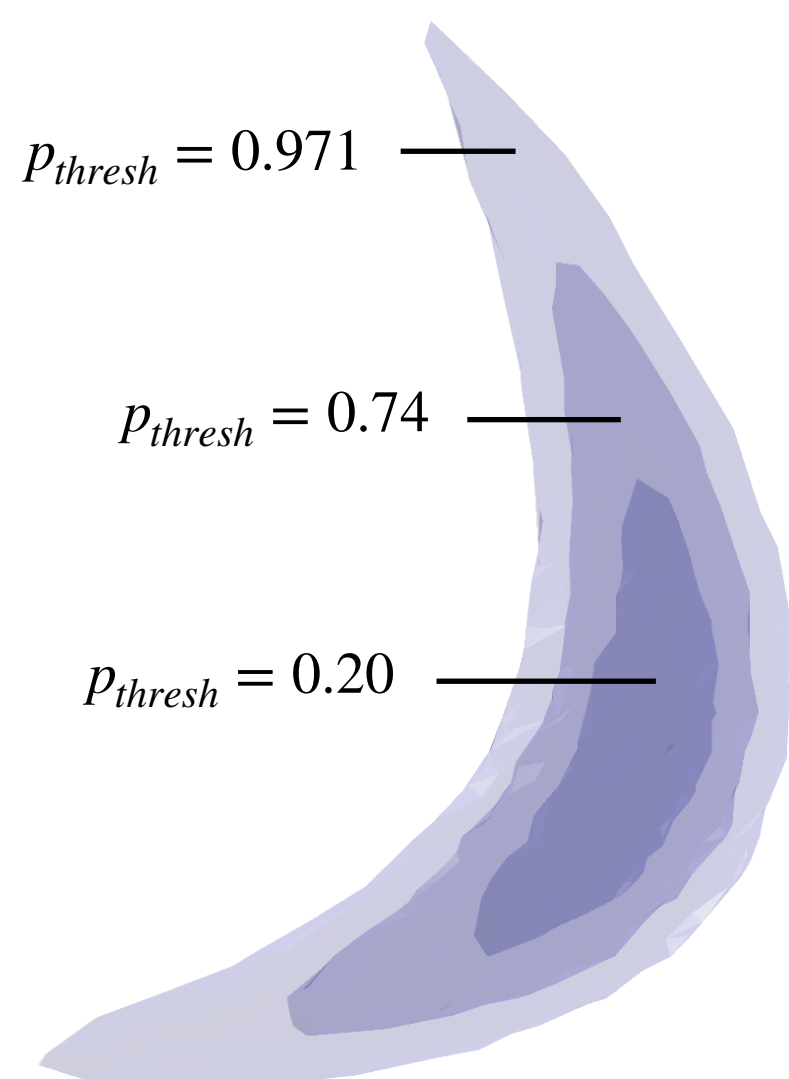


### $\alpha$ -Convex Hull Generation

where

$$p^* = \sum_{i=1}^M p_i \leq p_{thresh}$$

3D Isosurface  
Representation



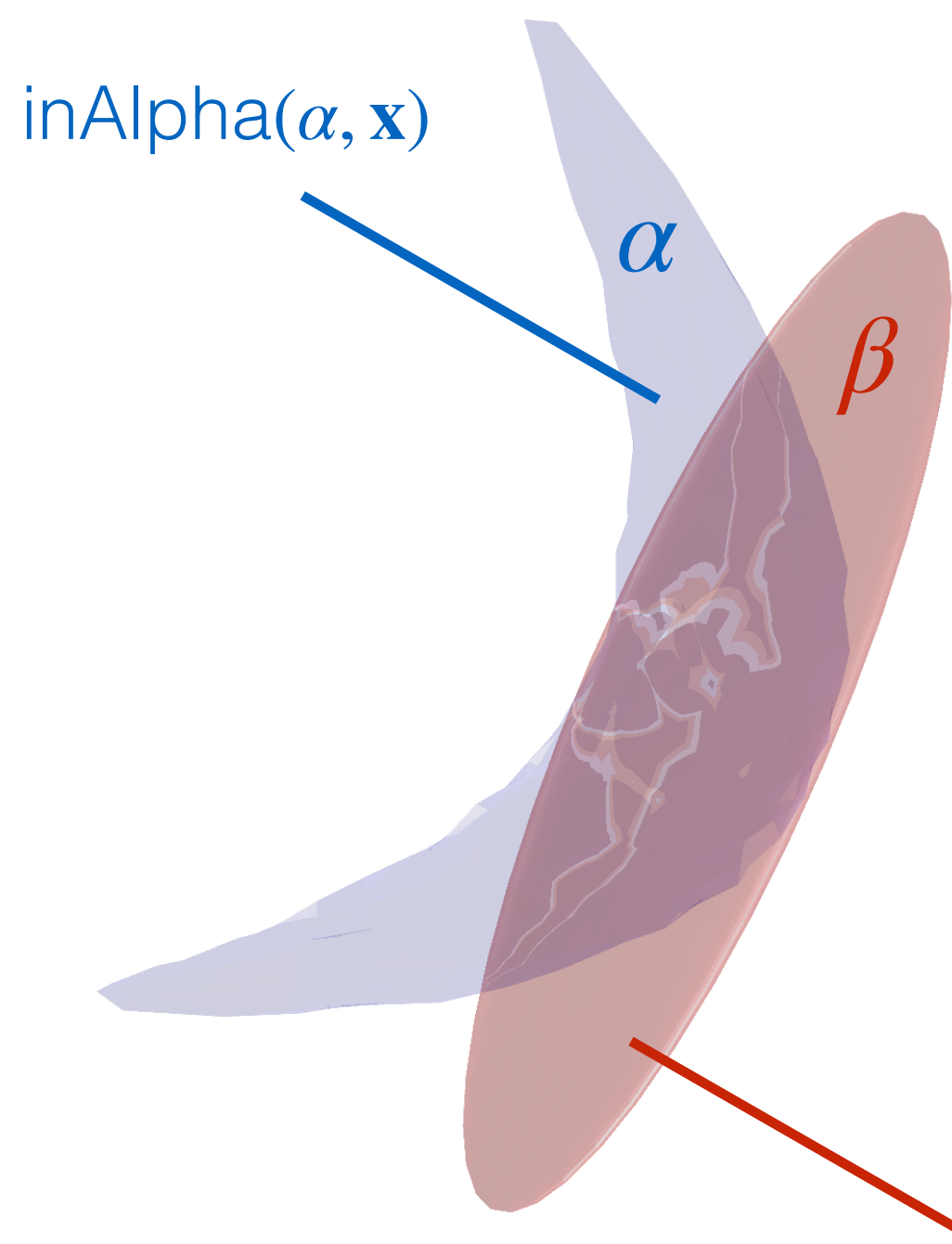
	$1\sigma$	$2\sigma$	$3\sigma$
1D	68%	95%	99.7%
2D	39%	86%	98.9%
3D	20%	74%	97.1%

Edelsbrunner, Herbert, et al. "Three-dimensional alpha shapes." ACM. (1994)

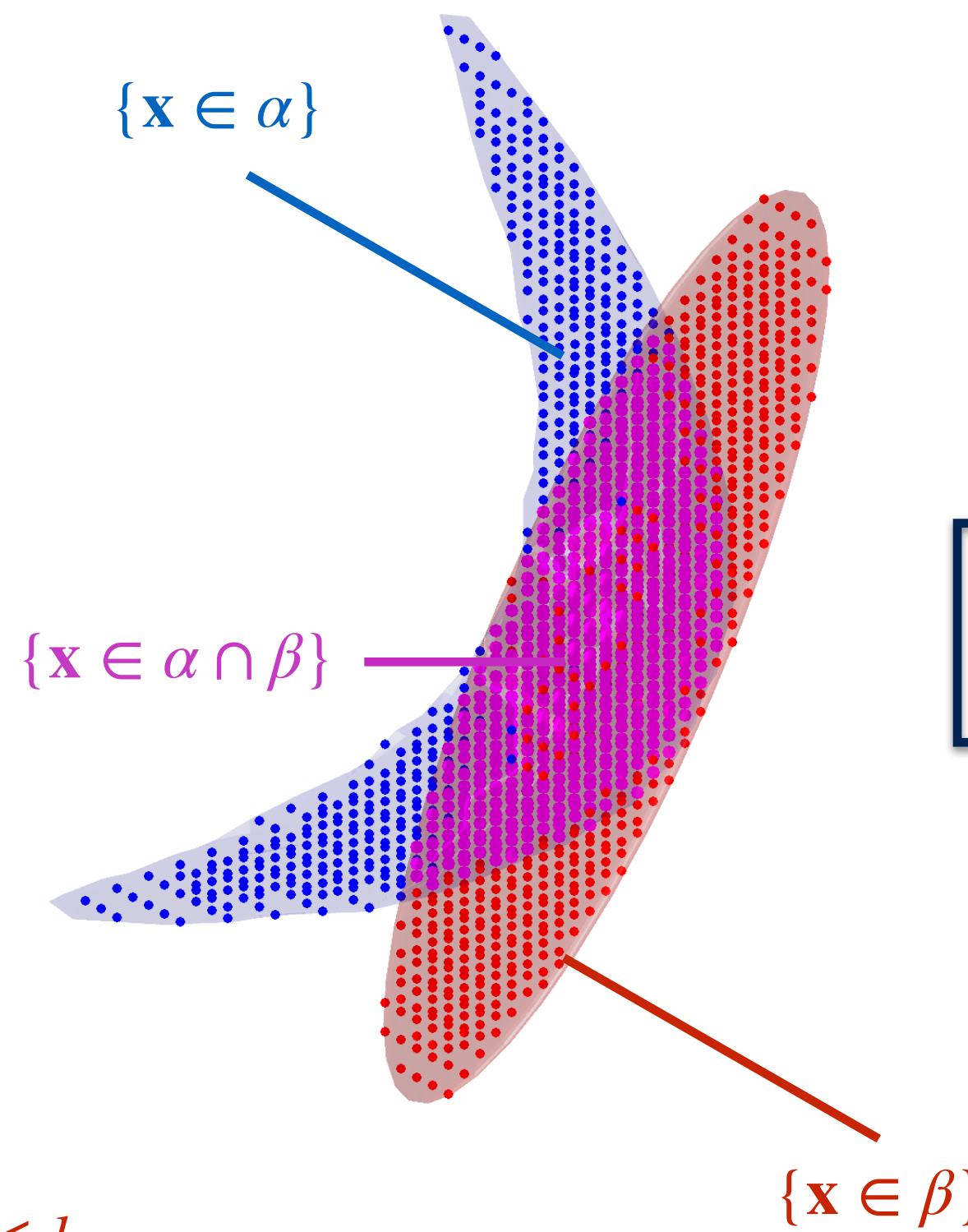


## Choosing a metric for Gaussian/non-Gaussian distribution comparison

### Distributions of interest



### Discretization



$$J(\alpha, \beta) \approx 0.39$$

Perfect match  $J = 1$

No overlap  $J = 0$



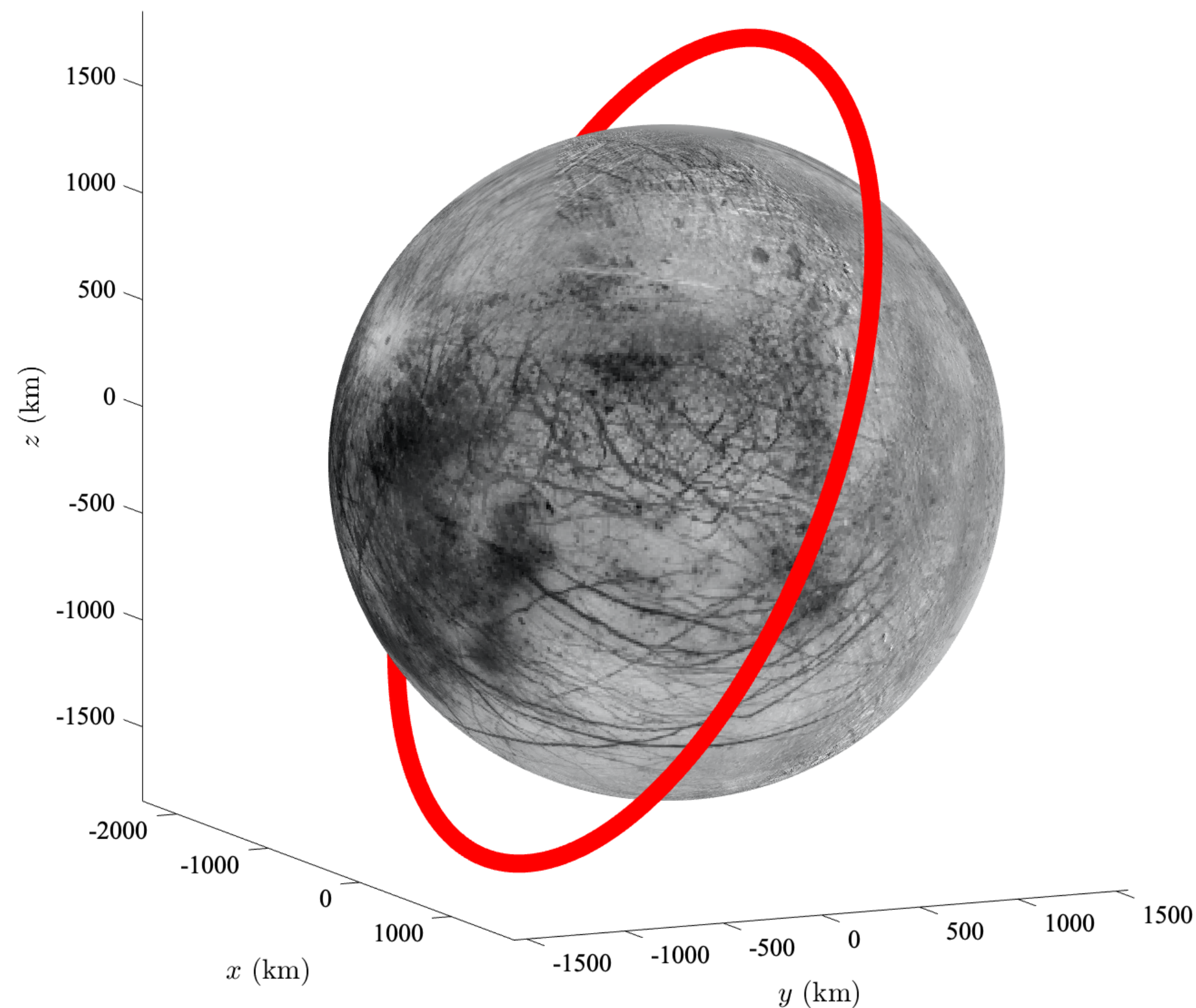
$$d_M = \sqrt{(\mathbf{x} - \mu)^T \Sigma^{-1} (\mathbf{x} - \mu)} \leq k$$

Jaccard Index  $\equiv J(\alpha, \beta) = \frac{|\{\mathbf{x} \in \alpha \cap \beta\}|}{|\{\mathbf{x} \in \alpha \cup \beta\}|} = \frac{|\{\mathbf{x} \in \alpha \cap \beta\}|}{|\{\mathbf{x} \in \alpha\}| + |\{\mathbf{x} \in \beta\}| - |\{\mathbf{x} \in \alpha \cap \beta\}|}$  where  $|\cdot|$  = size of set  $\mathbf{x} \in \mathbb{R}^3$



## Revised framework applied to measurement-sparse Jovian estimation

- Implement linear filter estimation with new comparison framework on Jovian trajectory:
  - \* Initial condition resulting in highly-inclined, low-Europa orbit
  - \* Propagated for 4 revolutions (11.279 hours) w/ RK8(7)
  - \* No measurements and negligible process noise
  - \*  $\alpha$ -convex hull comparison metric



$$\mathbf{x}_0 = \begin{bmatrix} a \text{ (km)} \\ e \text{ ( )} \\ i \text{ (}^\circ\text{)} \\ \Omega \text{ (}^\circ\text{)} \\ \omega \text{ (}^\circ\text{)} \\ M \text{ (}^\circ\text{)} \end{bmatrix} = \begin{bmatrix} 2029.4809 \\ 0.17 \\ 112.3^\circ \\ 180^\circ \\ 180^\circ \\ 0^\circ \end{bmatrix}$$

$$\sigma_r = 1 \text{ km}, \quad \sigma_v = 1 \text{ m/s}$$

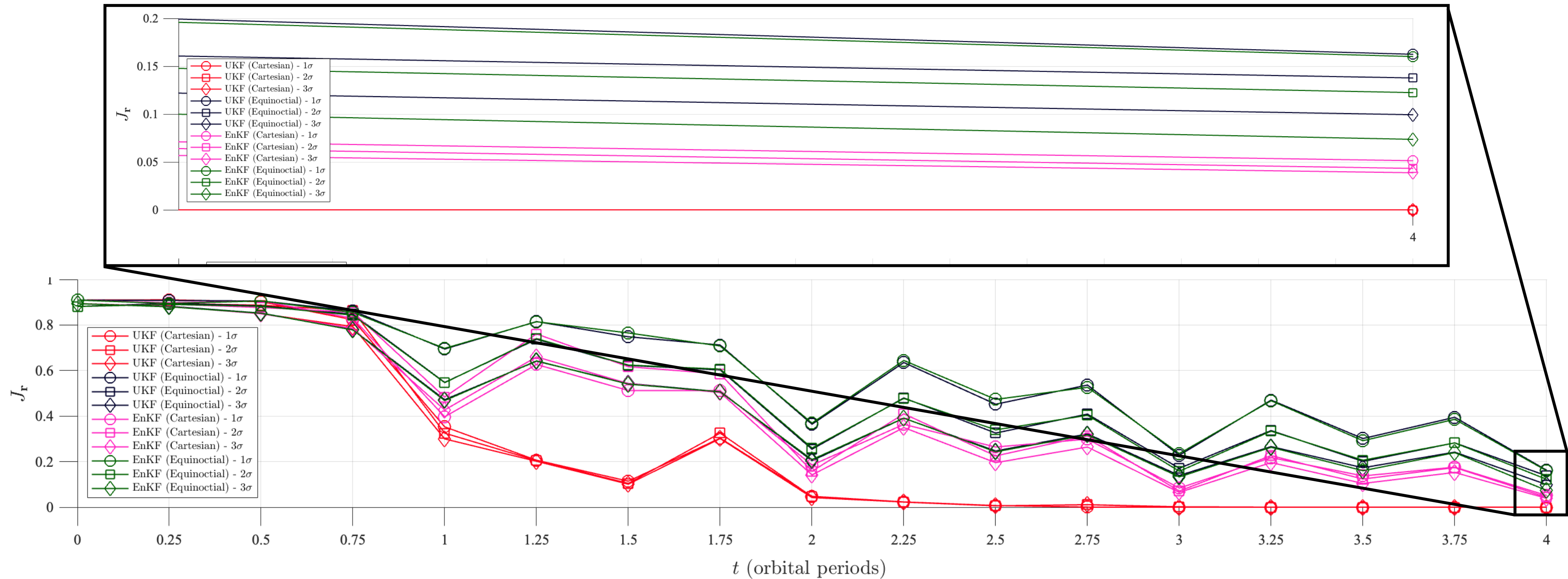
Filter	Parameters
Particle Filter (truth)	Particles: $10^5$
UKF	$\alpha = 10^{-3}, \beta = 2, \kappa = 0$
EnKF	Members: $10^4$





# Jovian Application: Low-Europa Orbit

## Evaluating the efficacy of linear filters for measurement-sparse estimation



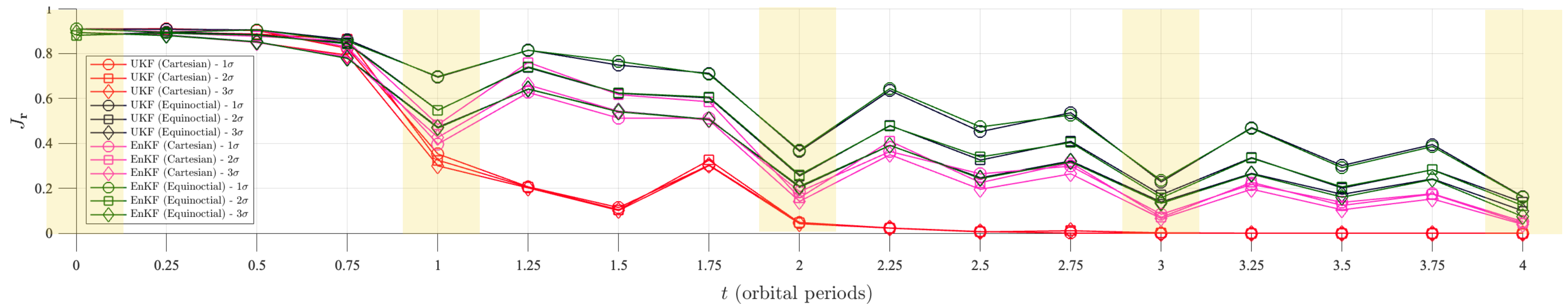
$J_r$	UKF (Cartesian)	UKF (Equinoctial)	EnKF (Cartesian)	EnKF (Equinoctial)
$1\sigma$	N/A	<b>0.1763</b>	0.0445	0.1727
$2\sigma$	N/A	<b>0.1414</b>	0.0427	0.1237
$3\sigma$	N/A	<b>0.1049</b>	0.0366	0.0800





# Jovian Application: Low-Europa Orbit

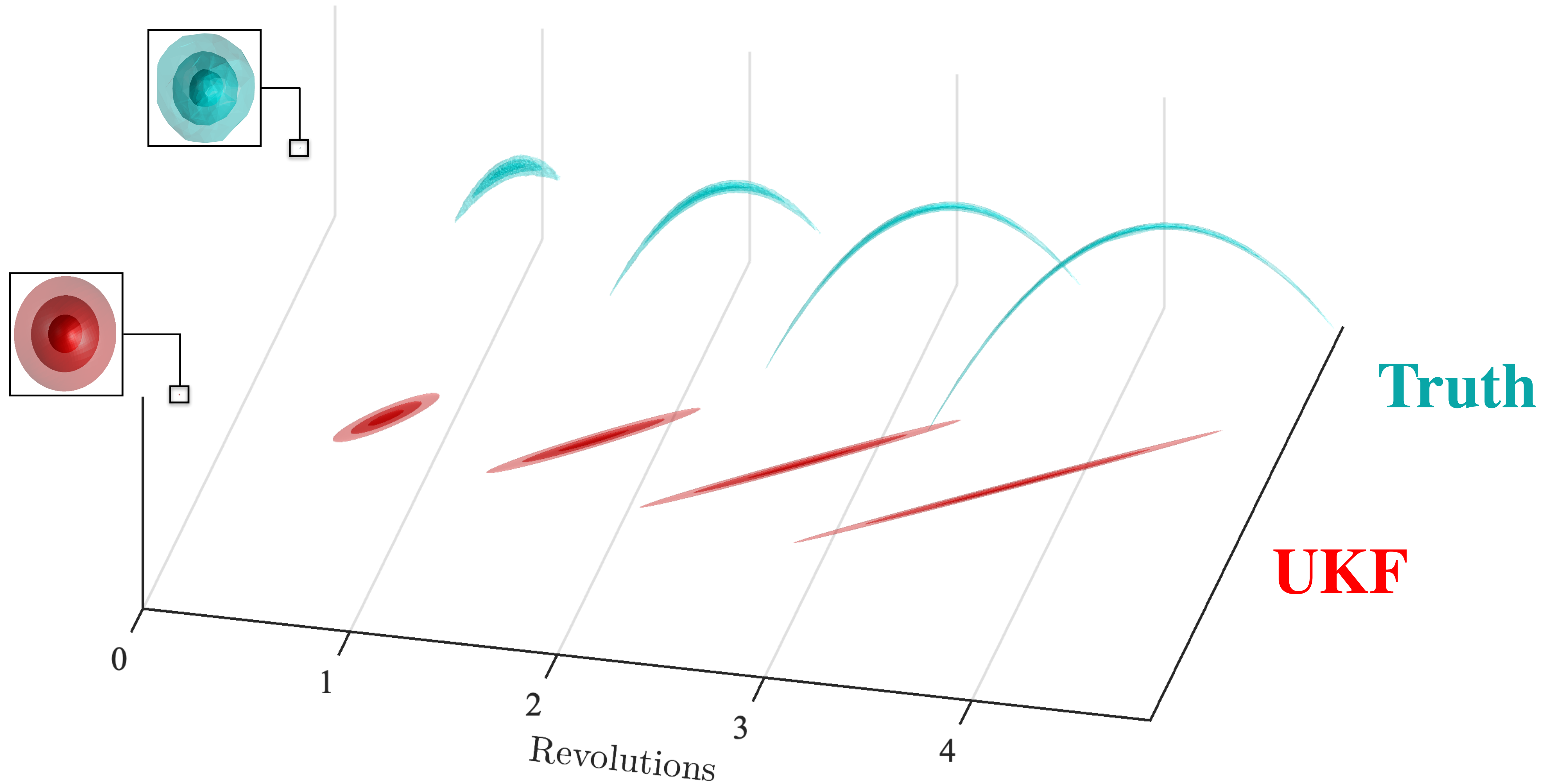
## Evaluating the efficacy of linear filters for measurement-sparse estimation





# Jovian Application: Low-Europa Orbit

Evaluating the efficacy of linear filters for measurement-sparse estimation

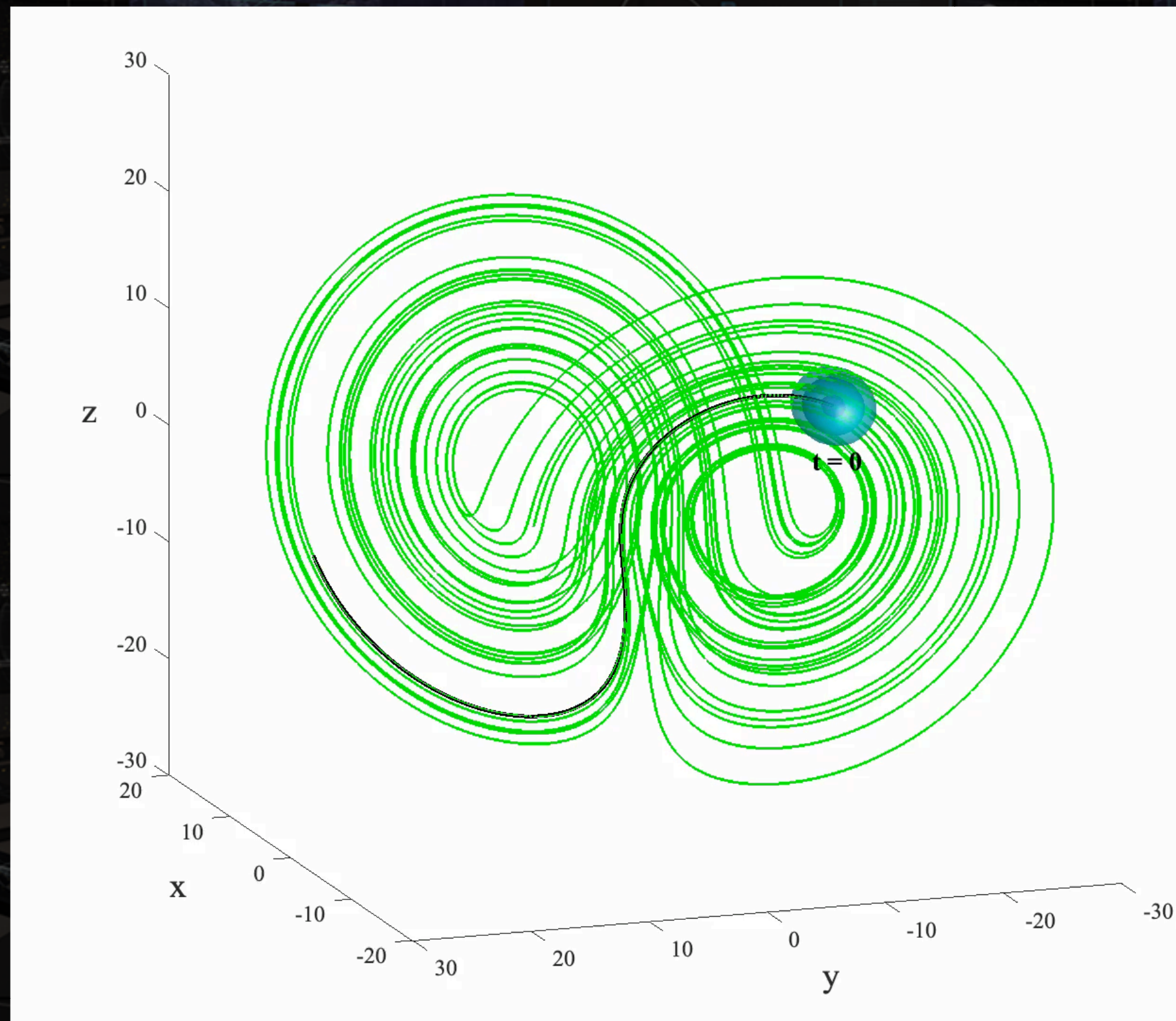




# Motivation for New Nonlinear Filters

## Addressing the shortcomings of the particle filter

- To address the shortcomings of the linear filter, we utilize...



## Grid-based Bayesian Estimation Exploiting Sparsity (GBEES)

An efficient Bayesian estimation method for representing and propagating uncertainty

GBEES is a 2nd-order accurate, Godunov finite volume method that treats probability as a fluid, flowing the PDF through phase space subject to the dynamics of the system. Because of its formulation, it can handle deterministic/stochastic systems while maintaining resolution.



## Grid-based Bayesian Estimation Exploiting Sparsity (GBEES)

- GBEES consists of two distinct processes, one performed in **continuous-time**, the other in **discrete-time**:

1. The probability distribution function  $p_{\mathbf{x}}(\mathbf{x}', t)$  is continuous-time marched via the **Fokker-Planck Equation**:

$$\frac{\partial p_{\mathbf{x}}(\mathbf{x}', t)}{\partial t} = - \frac{\partial f_i(\mathbf{x}', t) p_{\mathbf{x}}(\mathbf{x}', t)}{\partial x'_i} + \frac{1}{2} \frac{\partial^2 q_{ij} p_{\mathbf{x}}(\mathbf{x}', t)}{\partial x'_i \partial x'_j}$$

\*  $f_i$ : advection (EOMs) in the  $i^{\text{th}}$  dimension

\*  $q_{ij}$ :  $(i, j)^{\text{th}}$  element of the spectral density ( $Q = 0$ , PDE is hyperbolic)

2. At discrete-time interval  $t_k$ , measurement  $y_k$  updates  $p_{\mathbf{x}}(\mathbf{x}', t)$  via

**Bayes' Theorem**:

$$p_{\mathbf{x}}(\mathbf{x}', t_{k+}) = \frac{p_{\mathbf{y}}(\mathbf{y}_k | \mathbf{x}') p_{\mathbf{x}}(\mathbf{x}', t_{k-})}{C}$$

\*  $p_{\mathbf{x}}(\mathbf{x}', t_{k+})$ : a posteriori distribution

\*  $p_{\mathbf{y}}(\mathbf{y}_k | \mathbf{x}')$ : measurement distribution

\*  $p_{\mathbf{x}}(\mathbf{x}', t_{k-})$ : a priori distribution

\*  $C$ : normalization constant



## Grid-based Bayesian Estimation Exploiting Sparsity (GBEES)

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2. At discrete-time interval  $t_k$ , measurement  $y_k$  updates  $p_{\mathbf{x}}(\mathbf{x}', t)$  via **Bayes' Theorem**:

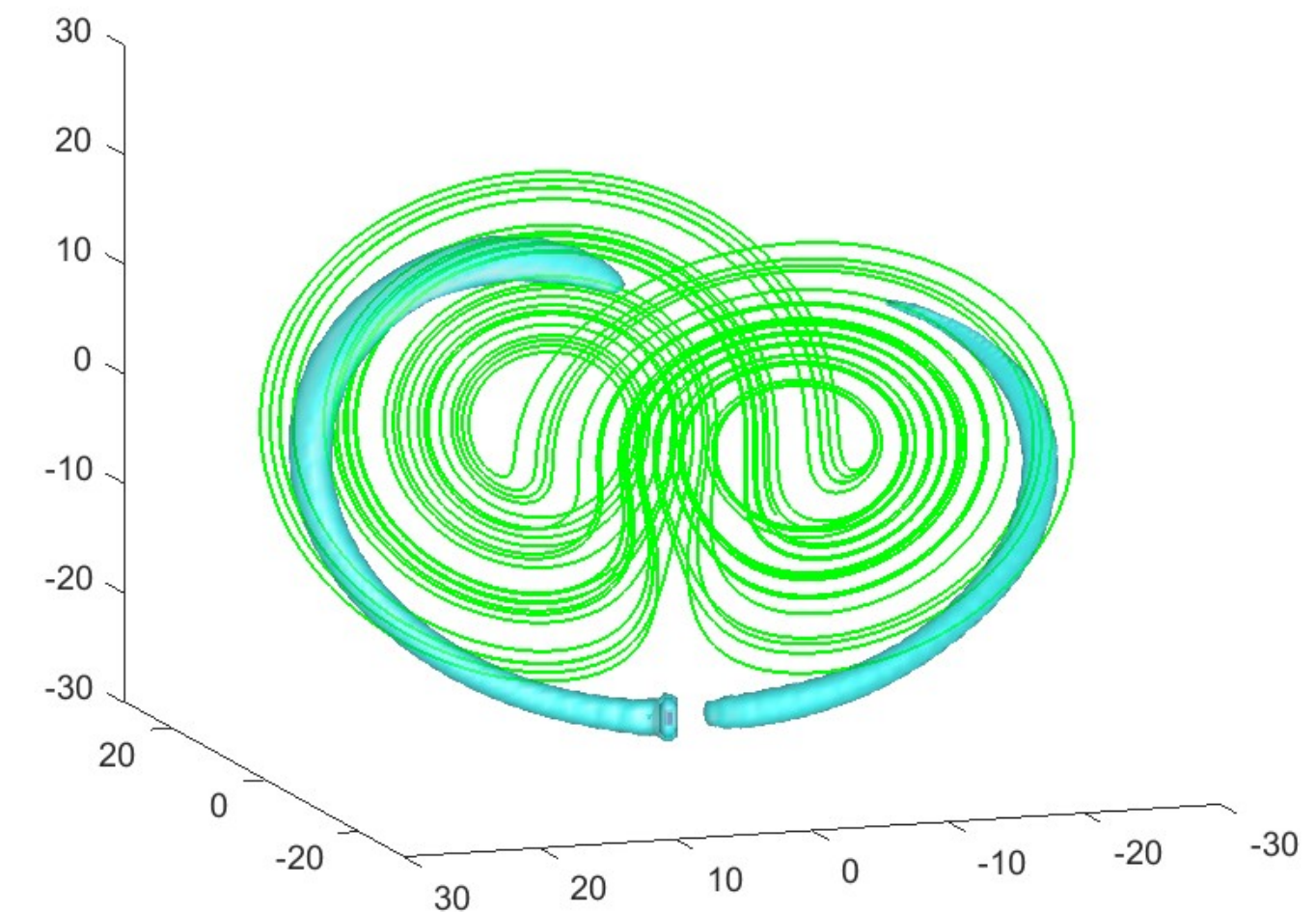
$$p_{\mathbf{x}}(\mathbf{x}', t_{k+}) = \frac{p_{\mathbf{y}}(\mathbf{y}_k | \mathbf{x}') p_{\mathbf{x}}(\mathbf{x}', t_{k-})}{C}$$

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**A priori**

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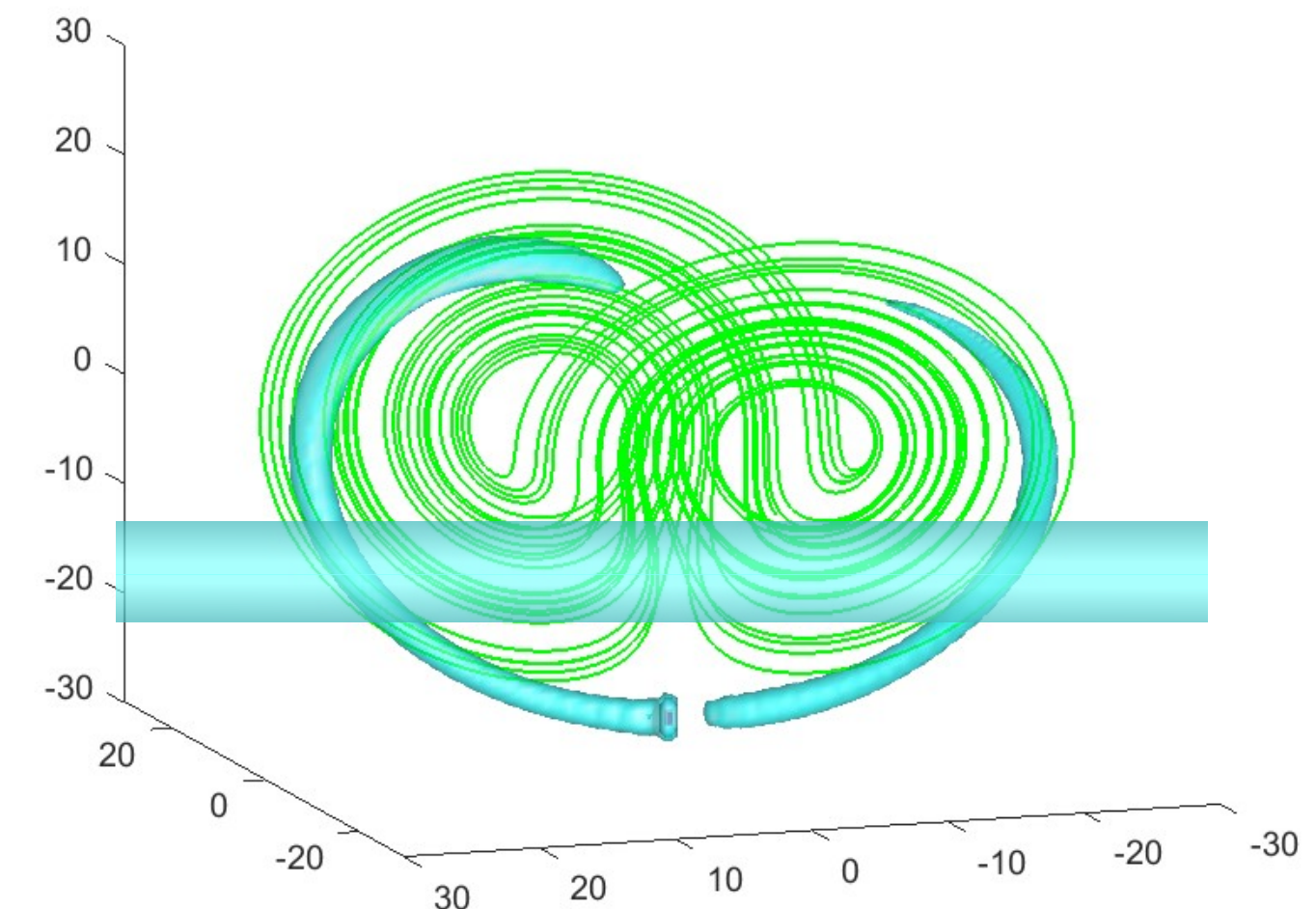
$$\frac{\partial p_{\mathbf{x}}(\mathbf{x}', t)}{\partial t} = - \frac{\partial f_i(\mathbf{x}', t) p_{\mathbf{x}}(\mathbf{x}', t)}{\partial x'_i} + \frac{1}{2} \frac{\partial^2 q_{ij} p_{\mathbf{x}}(\mathbf{x}', t)}{\partial x'_i \partial x'_j}$$

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- \*  $C$ : normalization constant



**A priori** × **Measurement**



## Grid-based Bayesian Estimation Exploiting Sparsity (GBEES)

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2. At discrete-time interval  $t_k$ , measurement  $y_k$  updates  $p_{\mathbf{x}}(\mathbf{x}', t)$  via **Bayes' Theorem**:

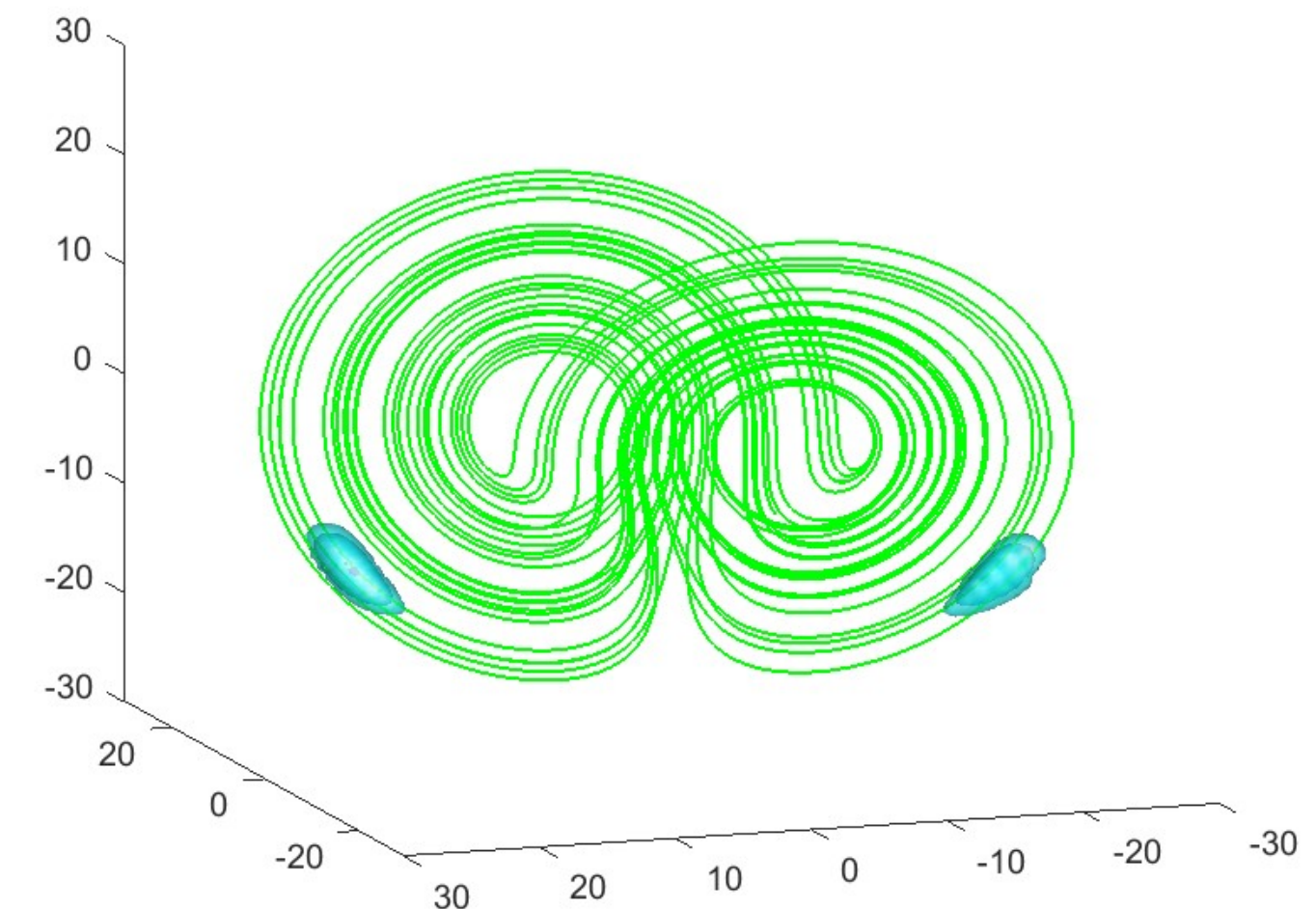
$$p_{\mathbf{x}}(\mathbf{x}', t_{k+}) = \frac{p_{\mathbf{y}}(\mathbf{y}_k | \mathbf{x}') p_{\mathbf{x}}(\mathbf{x}', t_{k-})}{C}$$

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\*  $C$ : normalization constant



**A priori** × **Measurement** = **A posteriori**

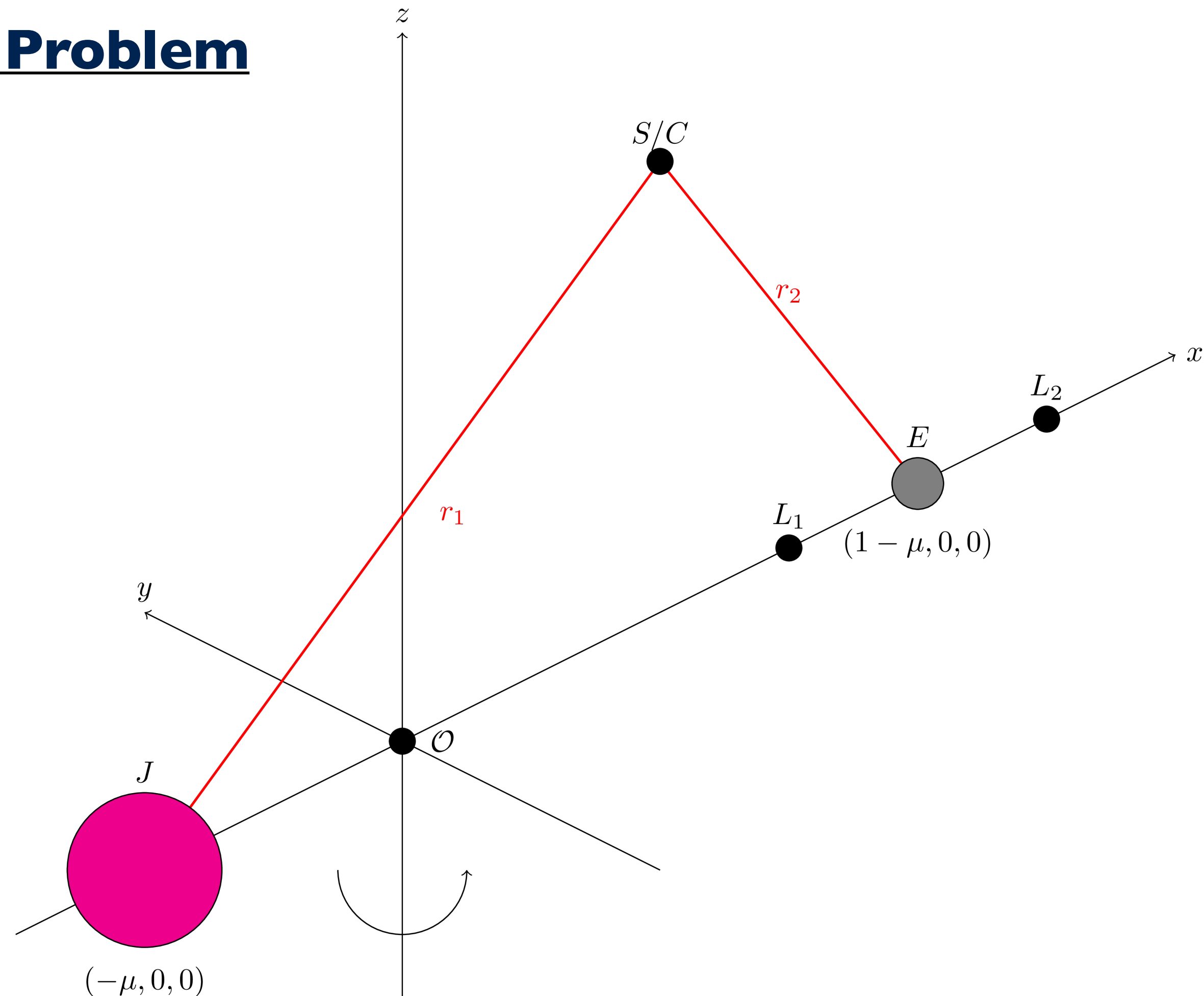
- We look to apply the developed framework to another systems applicable to Jovian trajectories

### Circular Restricted Three-Body Problem

$$\mathbf{x} = \begin{bmatrix} x \\ y \\ z \\ v_x \\ v_y \\ v_z \end{bmatrix}, \quad \dot{\mathbf{x}} = \begin{bmatrix} v_x \\ v_y \\ v_z \\ 2v_y + \Omega_x \\ -2v_x + \Omega_y \\ \Omega_z \end{bmatrix}$$

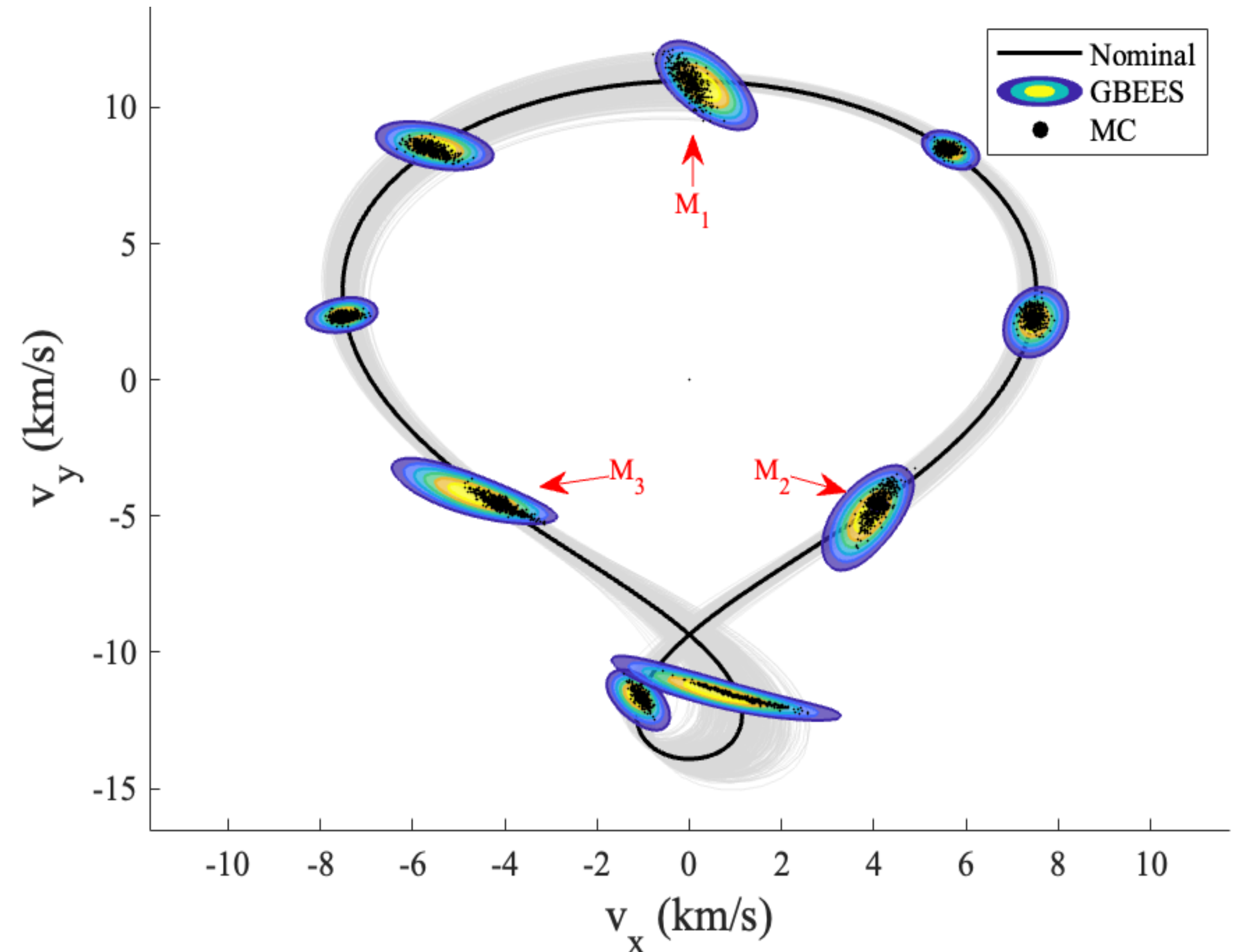
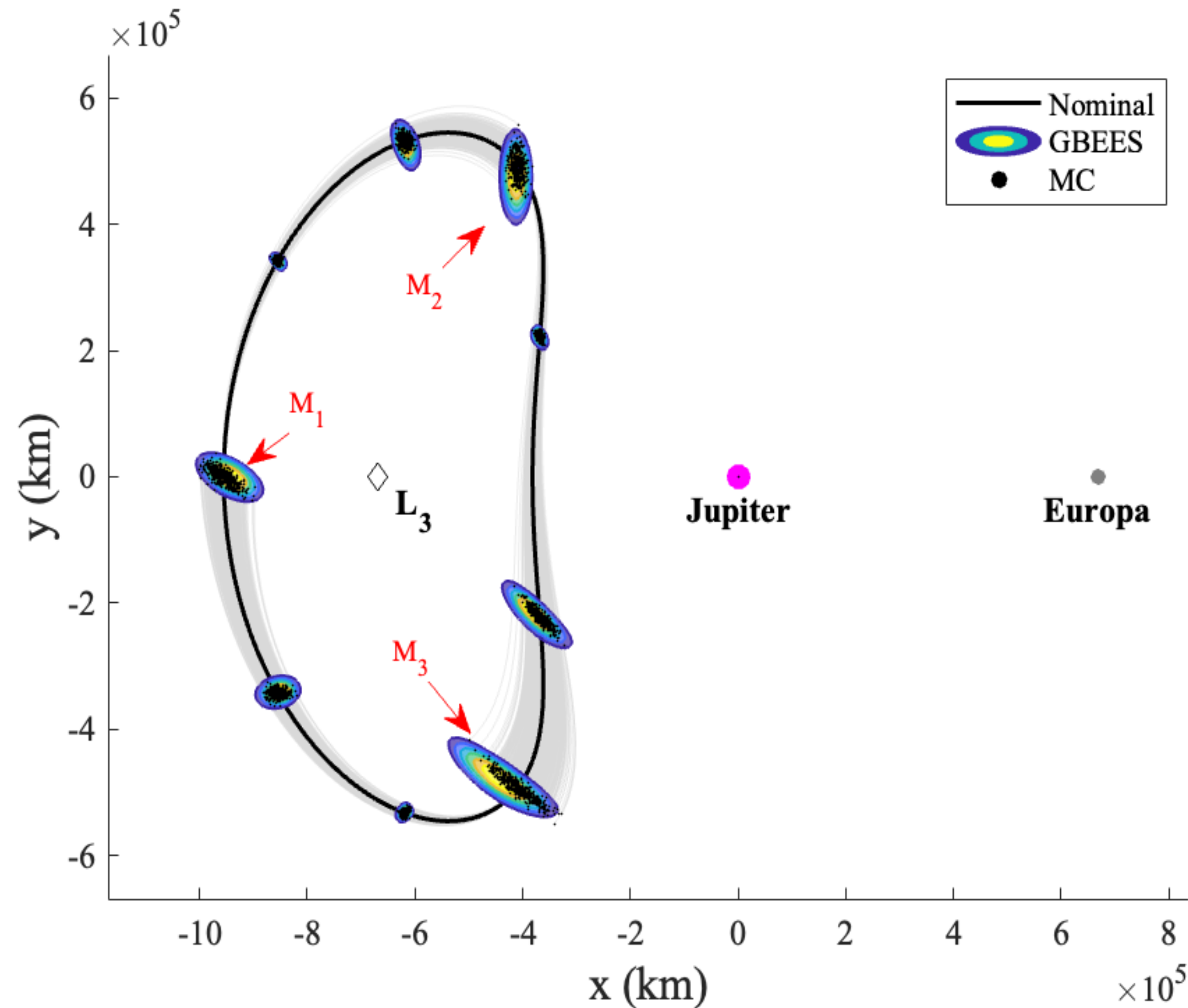
$$\text{where } \Omega(x, y) = \frac{x^2 + y^2}{2} + \frac{1 - \mu}{r_1} + \frac{\mu}{r_2} + \frac{\mu(1 - \mu)}{2}$$

- \* We use initial conditions generated from the JPL Three-Body Periodic Orbit Catalog for the Jupiter-Europa system





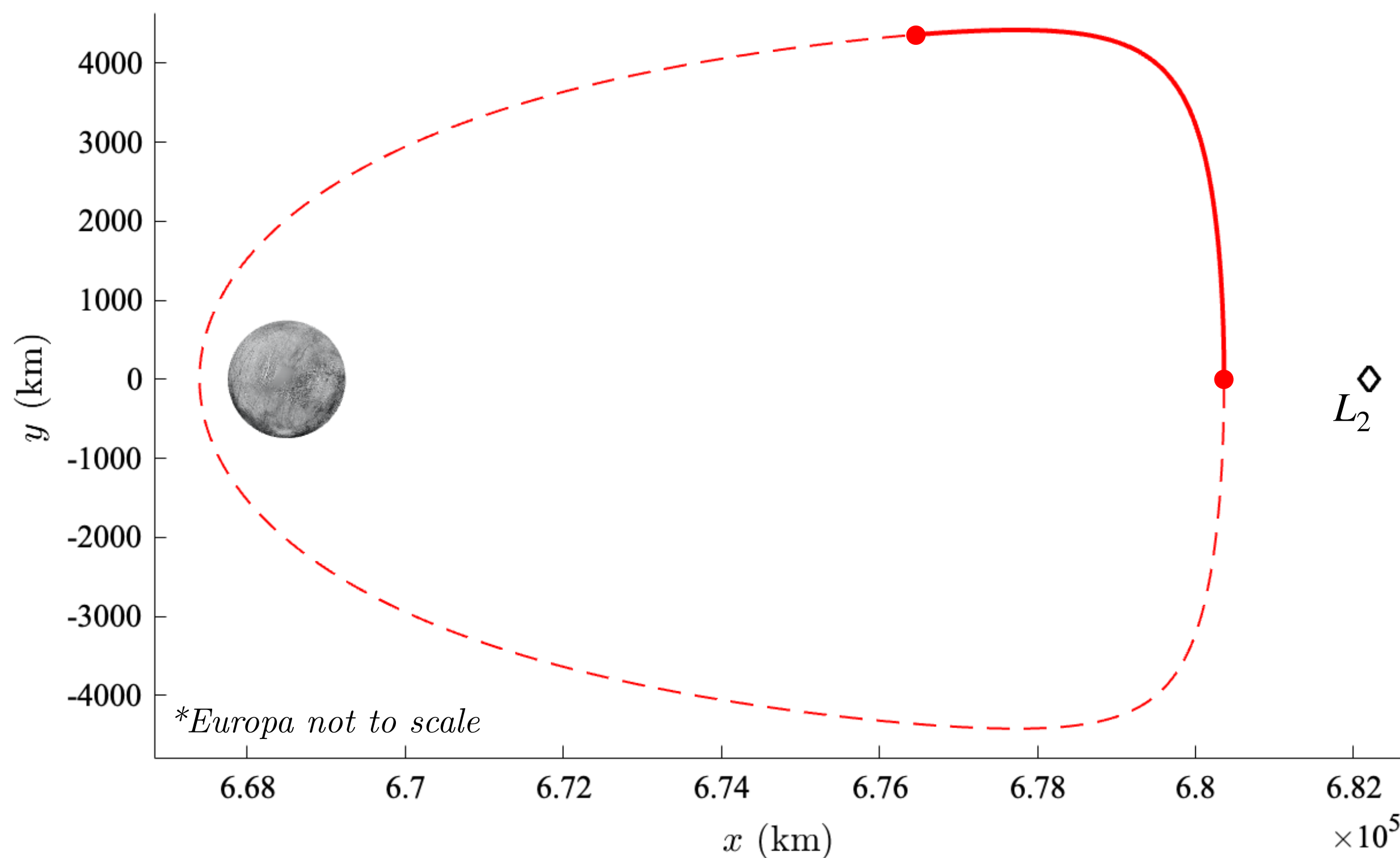
- Previous work applied a similar framework to **planar Lyapunov orbits** about  $L_3$  in the Jupiter-Europa 3BP



- We found that uncertainty remained near **Gaussian**, even with an infrequent measurement cadence ( $\sim 1.17$  days)

## Revised framework applied to measurement-sparse Jovian estimation

- Implement linear filter estimation with new comparison framework on Jovian trajectory:
  - \* Initial condition resulting in eastern, low-prograde orbit about Europa
  - \* Propagated for 14 hours w/ RK8(7) and GBEES
  - \* No measurements and negligible process noise
  - \*  $\alpha$ -convex hull comparison metric



$$\mathbf{x}_0 = \begin{bmatrix} x \text{ (km)} \\ y \text{ (km)} \\ v_x \text{ (m/s)} \\ v_y \text{ (m/s)} \end{bmatrix} = \begin{bmatrix} 6.803 \times 10^5 \\ 0 \\ 0 \\ 0.8623 \end{bmatrix}$$

$$\sigma_r = 100 \text{ km}, \quad \sigma_v = 10 \text{ m/s}$$

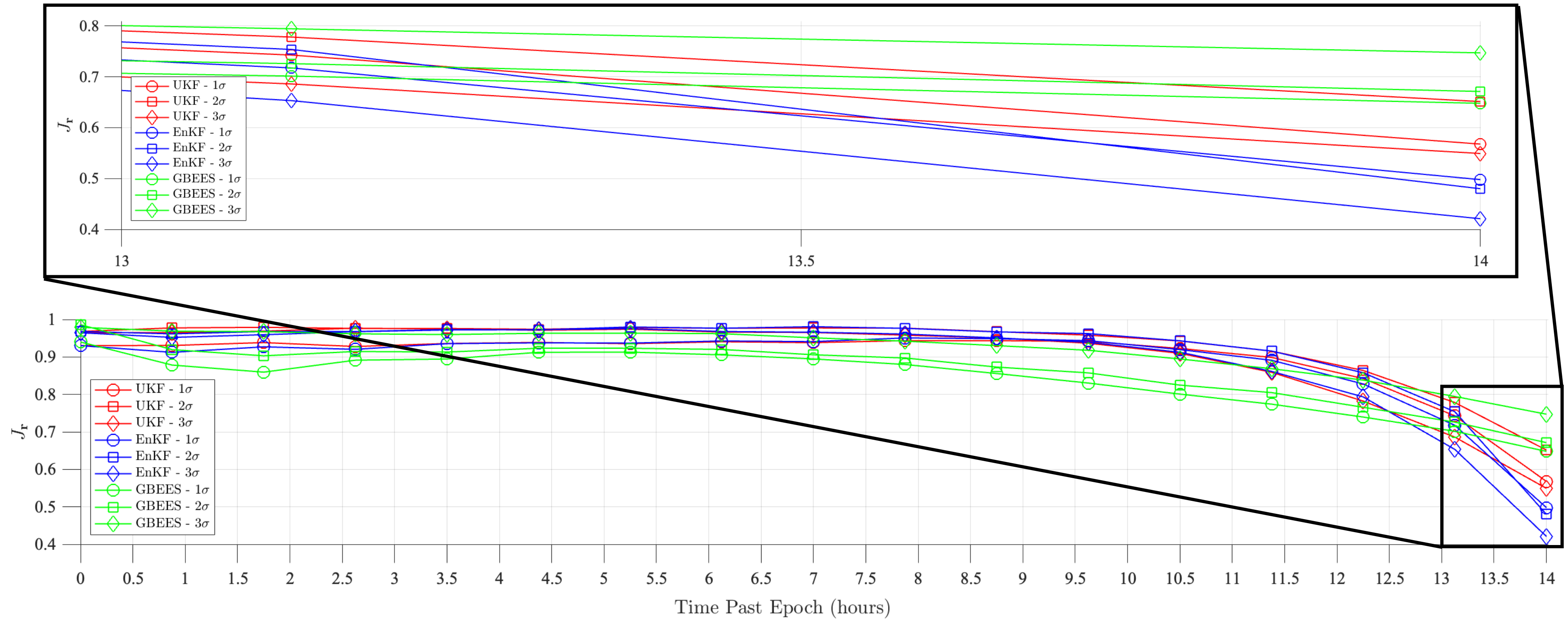
Filter	Parameters
Particle Filter (truth)	Particles: $10^6$
UKF	$\alpha = 10^{-3}, \beta = 2, \kappa = 0$
EnKF	Members: $10^4$
GBEES	$P_{thresh} = 10^{-7}$





# Jovian Application: Low-Prograde Orbit

## Comparing linear estimation with GBEEES

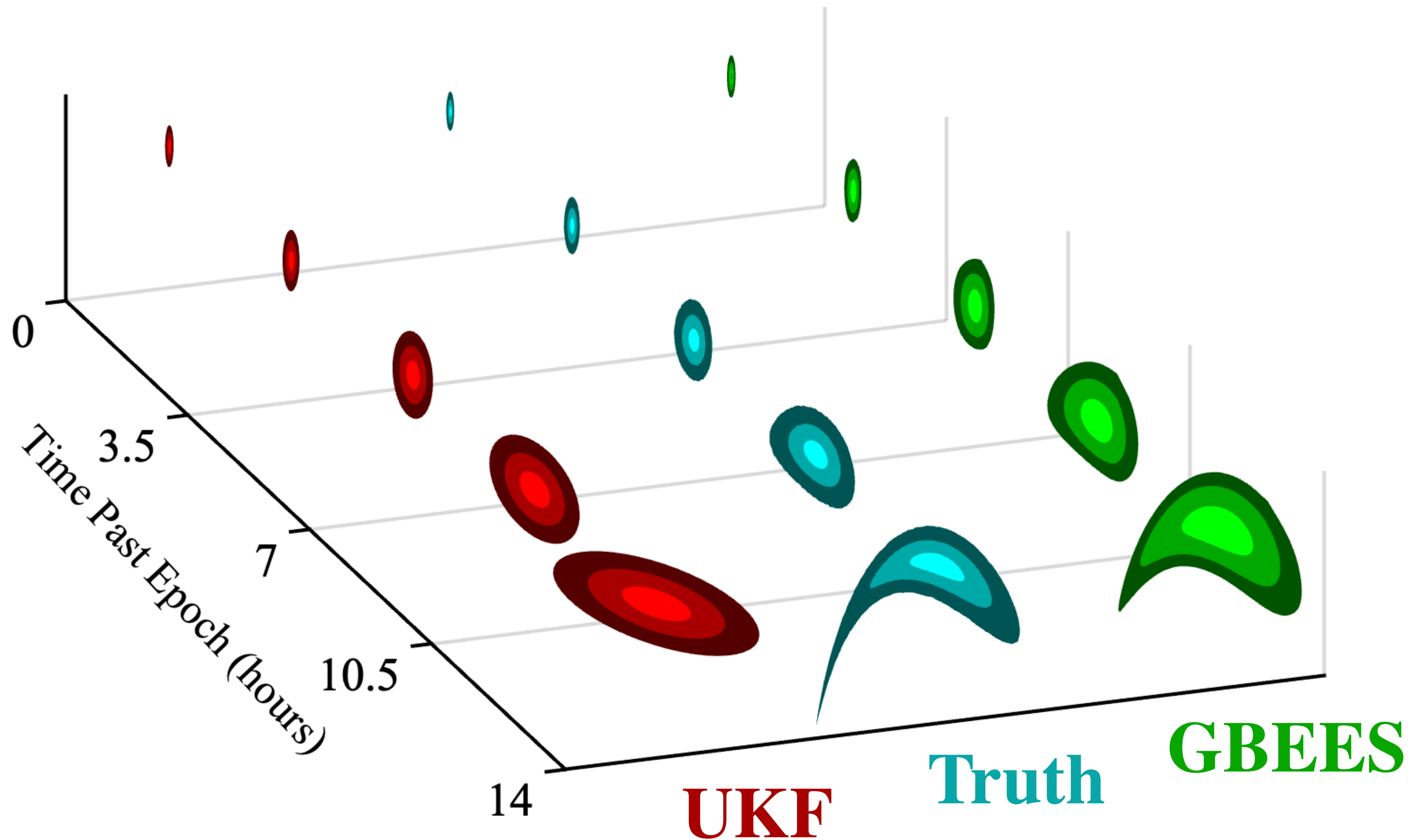


$J_r$	UKF	EnKF	GBEES
$1\sigma$	0.5678	0.4978	<b>0.6479</b>
$2\sigma$	0.6514	0.4800	<b>0.6713</b>
$3\sigma$	0.5492	0.4209	<b>0.7472</b>



# Jovian Application: Low-Prograde Orbit

Comparing linear estimation with GBEEs

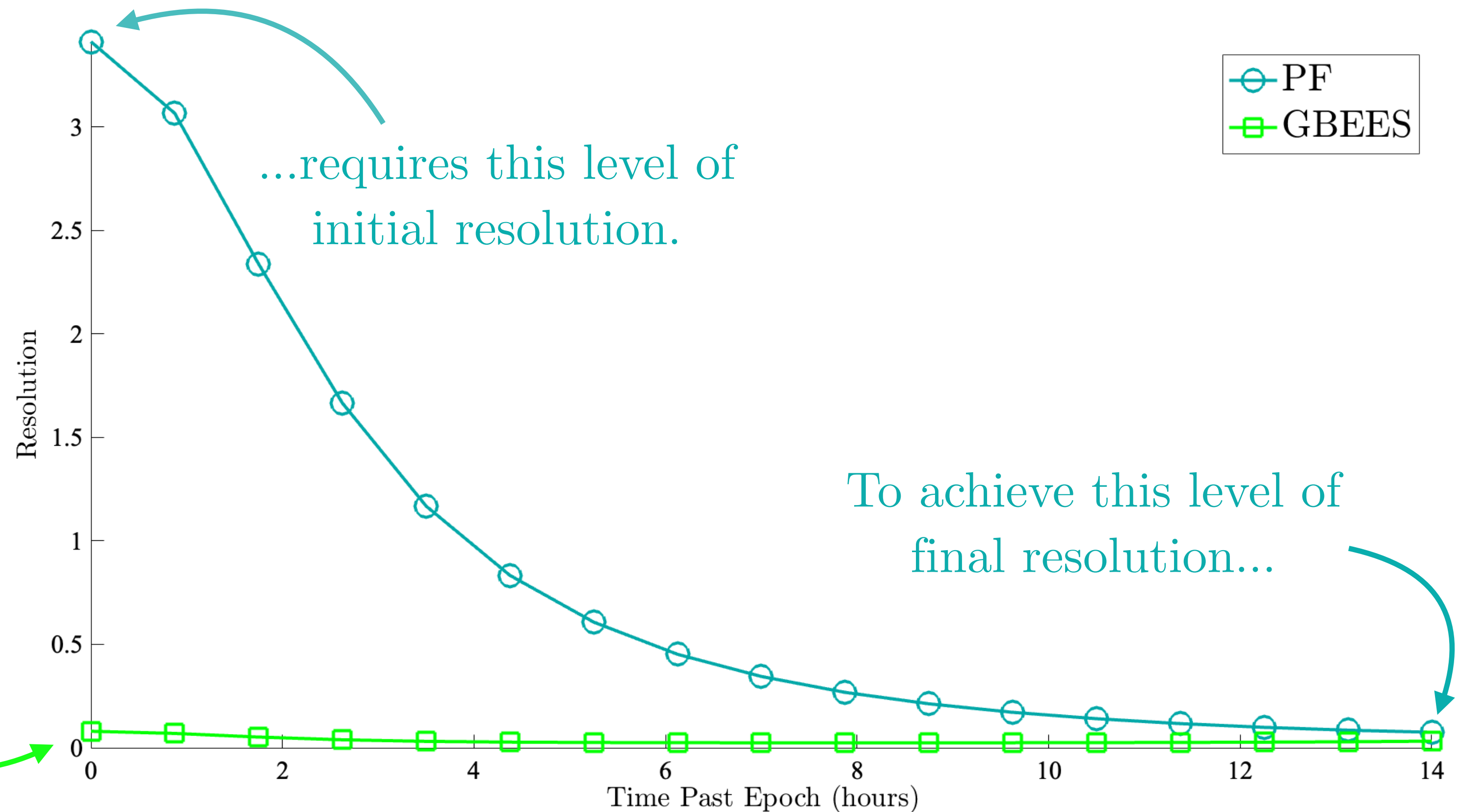




- We utilize a high-resolution PF as a truth distribution, so why don't we use it for estimation?
  - \* To achieve **sufficient resolution** at a distant measurement epoch requires a large (**usually unknown**) number of particles that are marched from the previous epoch

$$\text{Resolution} = \frac{\# \text{ of particles/grid cells}}{\text{Volume of uncertainty}}$$

Not the case for  
GBEES!



- \* GBEES nearly maintains resolution by **growing with the uncertainty**



# Conclusion

## Comments on Results and Future Work

### • Low-Europa Orbit

$J_r$	UKF (Cartesian)	UKF (Equinoctial)	EnKF (Cartesian)	EnKF (Equinoctial)
$1\sigma$	N/A	<b>0.1763</b>	0.0445	0.1727
$2\sigma$	N/A	<b>0.1414</b>	0.0427	0.1237
$3\sigma$	N/A	<b>0.1049</b>	0.0366	0.0800

\*  $1\sigma$  position uncertainty estimated by the UKF (Equinoctial) is able to maintain  $J_r \geq 0.5$  compared with truth distribution for nearly 2 revolutions without measurements, with local minima located at periapsis

### • Low-Prograde Orbit in Jupiter-Europa Three-Body System

$J_r$	UKF	EnKF	GBEES
$1\sigma$	0.5678	0.4978	<b>0.6479</b>
$2\sigma$	0.6514	0.4800	<b>0.6713</b>
$3\sigma$	0.5492	0.4209	<b>0.7472</b>

\* While linear filters are able to estimate uncertainty better when distributions are near-Gaussian, GBEES is more accurate when distributions are far from Gaussian, which occurs in about 14 hours for the given LPO

### • Future Work

- \* Propagating in the **slow-changing**, three-body local orbit elements
- \* **Parallelization** of Riemann solver embedded within GBEES
- \* Dynamics sourced from an **ephemeris-level** numerical integrator





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**All code can be found at: <https://github.com/bhanson10/GBEES> and <https://github.com/bhanson10/KePASSA2024>**

**Thank you for your time. Questions?**